SHATT AL-ARAB UNIVERSITY COLLEGE ENG. DEPARTMENT CIVIL ENG. STRENGTH OF MATERIALS -I SECOND YEAR DR. JASIM AL-BATTAT

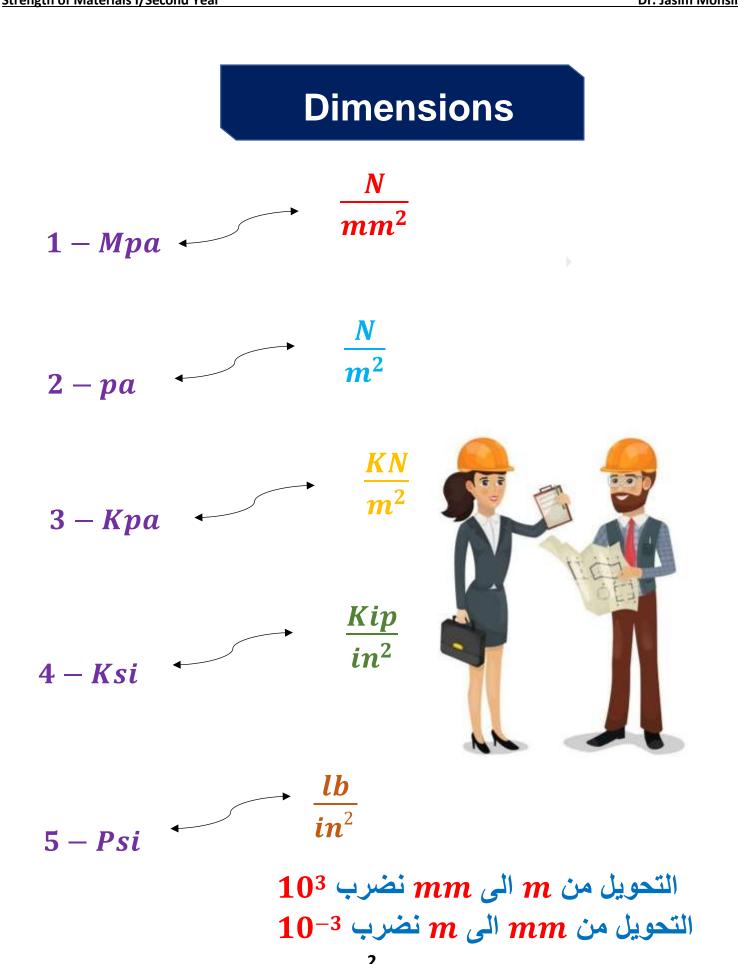
The Syllabus

- 1. stress
- 2. shear stress
- 3. strain
- 4. Torsion
- 5. thermal stress
- 6. DRAW SFD BMD
- 7. thin walled cylinder

References

- 1. Hibbeler Mechanics of Materials 8th Edition
- 2. Strength of Materials Andrew Pytel , Ferdinand L. Singer 3rd edition
- 3. Strength of Materials, Part 1, Elementary theory and problems
- 4. Lectures by Engineer Mustafa Jasim

Strength of Materials I/Second Year



Strength of Materials I/Second Year

Dr. Jasim Mohsin

التحويل من N الى KN نضرب 3-10 التحويل من KN الى N نضرب 10³

> التحويل من ft الى iN نضرب 12 التحويل من in الى ft نقسم 12

التحويل من Ksi اللي Psi نضرب 10³ التحويل من Psi اى Ksi نضرب 10⁻³



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1. Chapter one: Stress

1.1- Introduction

Mechanics of materials is a branch of mechanics that studies the internal effects of stress and strain in a solid body that is subjected to an external loading. Stress is associated with the strength of the material from which the body is made, while strain is a measure of the deformation of the body. In addition to this, mechanics of materials includes the study of the body's stability when a body such as a column is subjected to compressive Loading. A thorough understanding of the fundamentals of this subject is of vital importance because many of the formulas and rules of design cited in engineering codes are based upon the principles of this subject.

1.2- Equilibrium of a Deformable Body

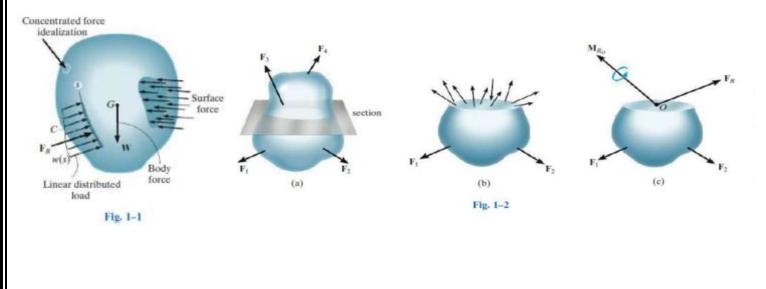
Since statics has an important role in both the development and application of mechanics of materials, it is very important to have a good grasp of its fundamentals. For this reason, we will review some of the main principles of statics that will be used throughout the text.

External Loads. A body is subjected to only two types of external loads; namely, Surface forces and body forces, Fig. 1-1.

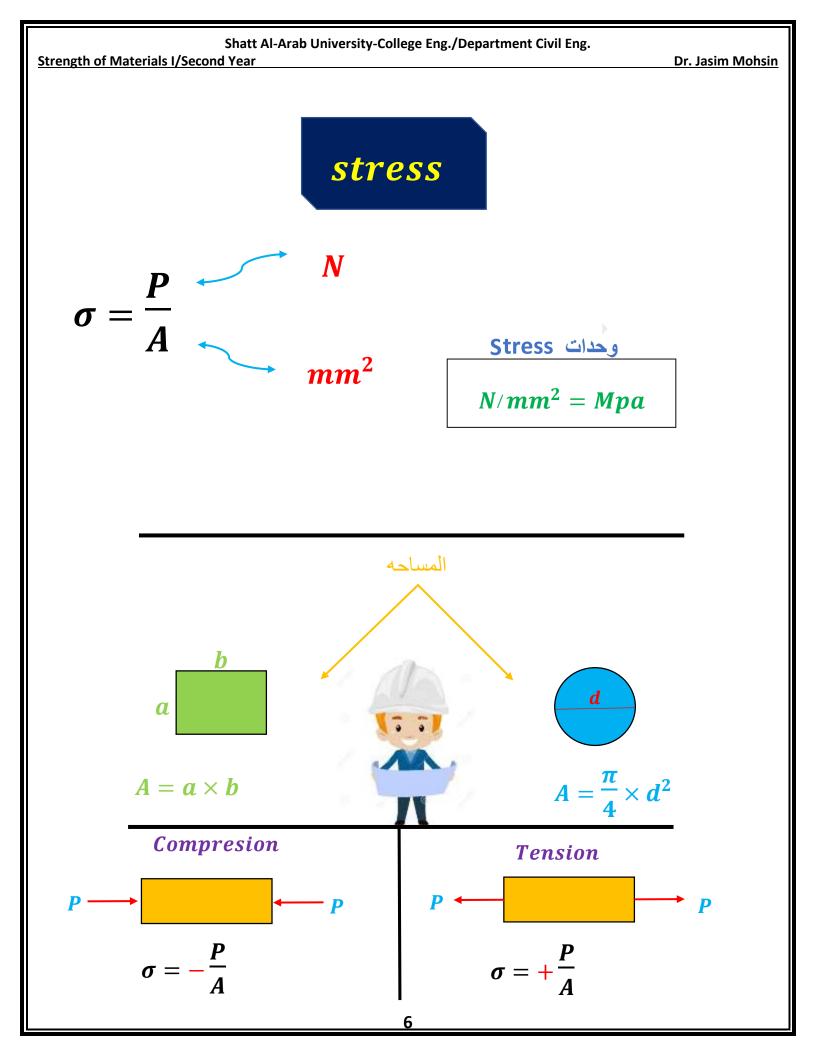
-Surface Forces. Surface forces are caused by the direct contact of one body with the surface of another.

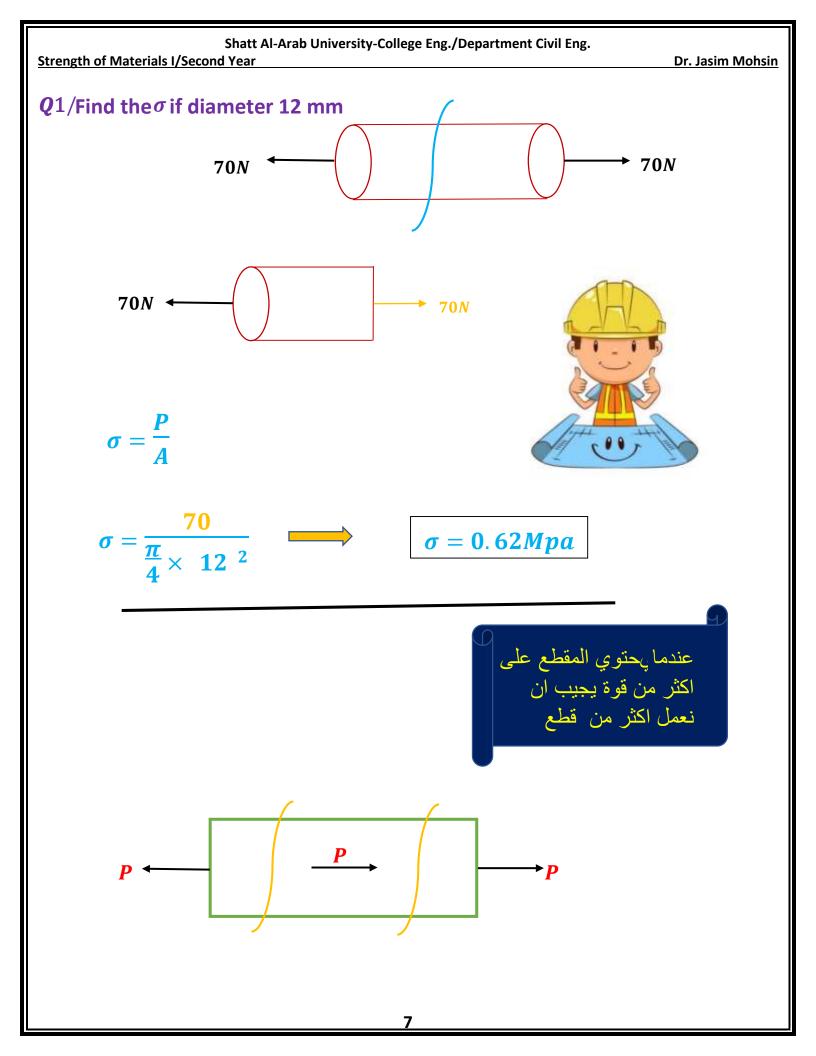
-Body Forces. A both' force is developed when one body exerts a force on another body without direct physical contact between the bodies. Examples include the effects caused by the earth's gravitation or its electromagnetic field.

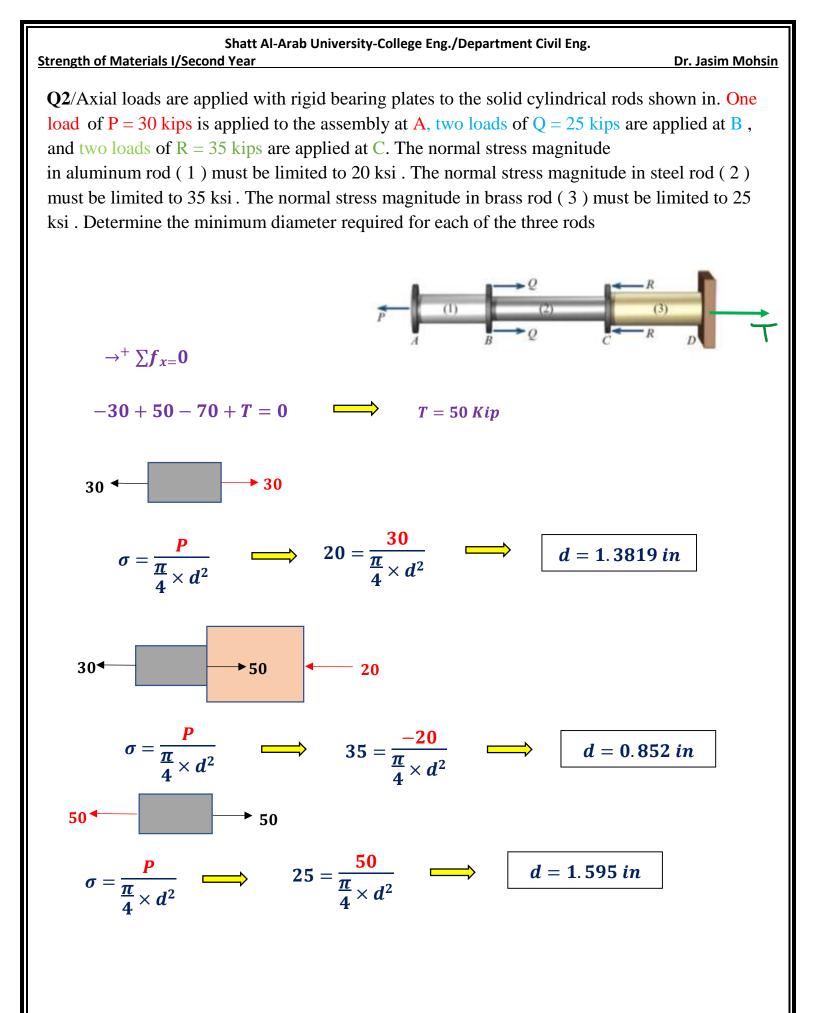
Internal Resultant Loadings. In mechanics of materials, statics is primarily used to determine the resultant loadings that act within a body. shown in Fig. 1-2



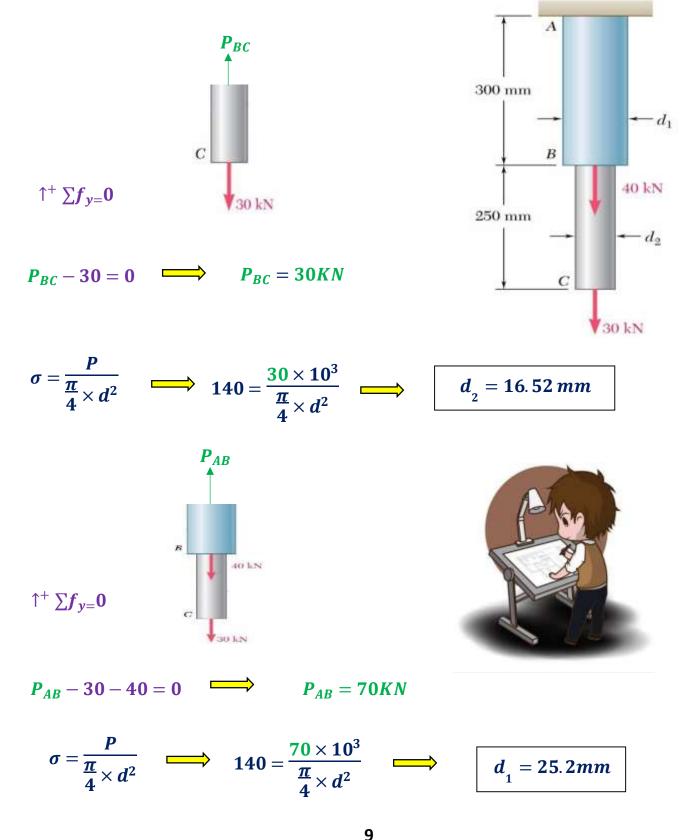
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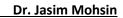




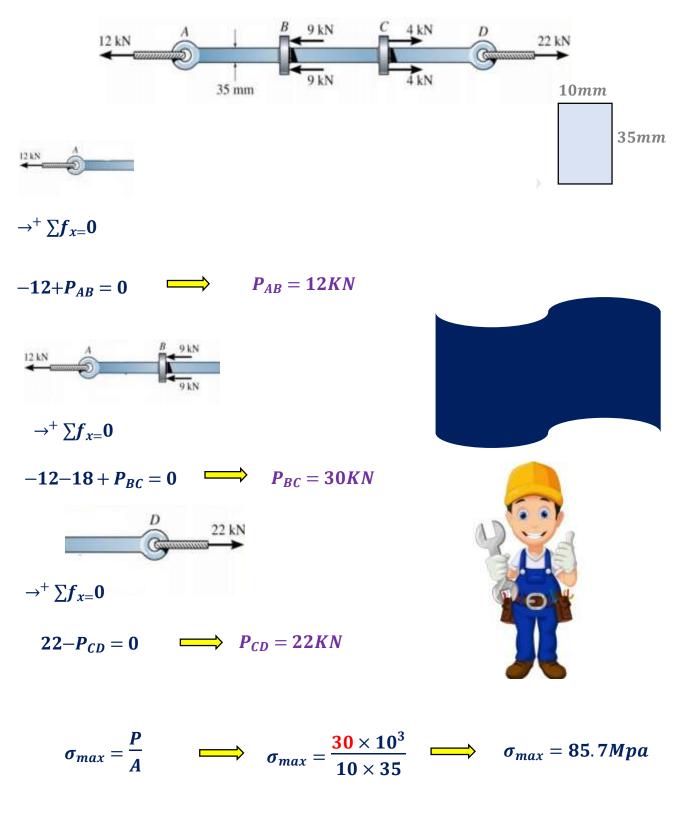


 ${\bf Q3}/{\rm Two}$ solid cylindrical rods AB and BC are welded together at B loaded as shown . Known that the average normal stress must not exceed 140 MPa in each rod . Determine the smallest allowable value of d1 , and d2





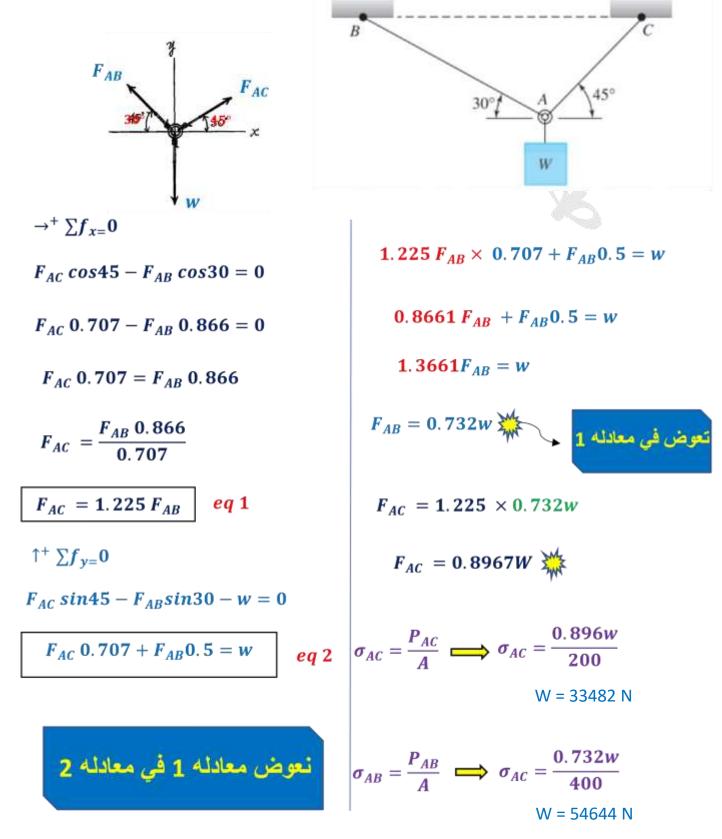
Q4/ has a constant width of 35 mm and a thickness of 10 mm . Determine the maximum average normal stress in the bar when it is subjected to the loading shown



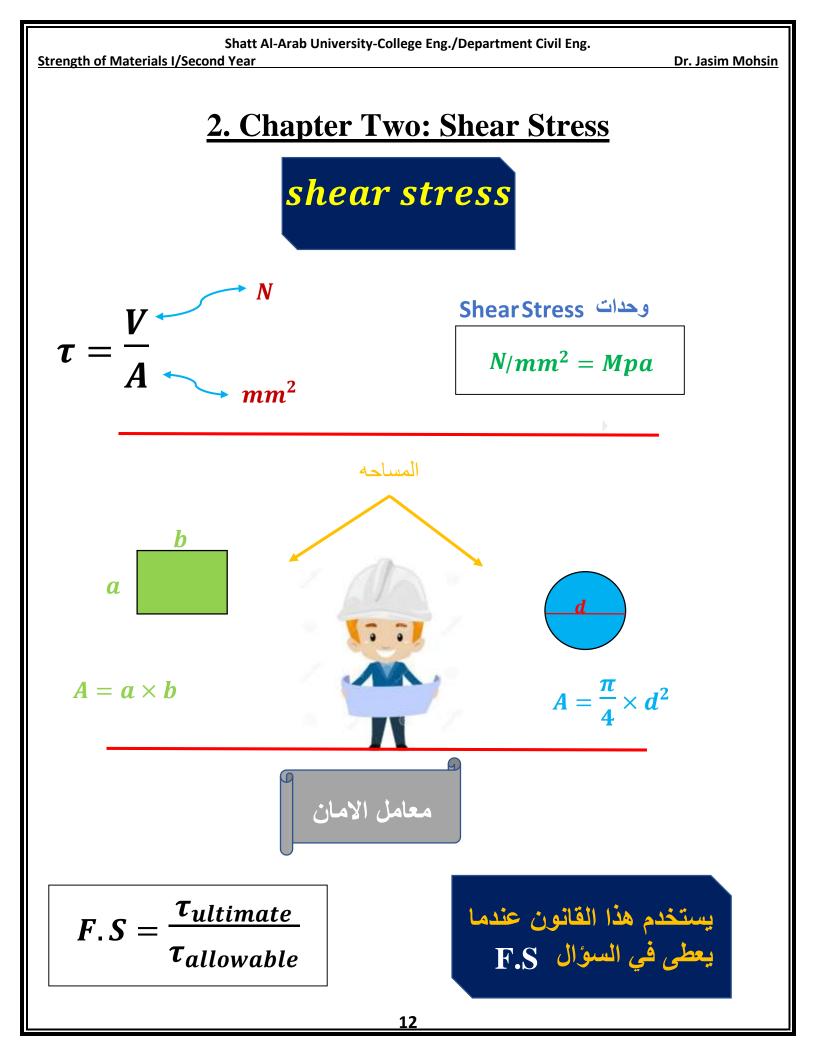
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<u>Q5</u> Determine the largest weight W that can be supported by the two wires AB and AC. The working stresses are 100 MPa for AB and 150 MPa for AC. The cross sectional areas of AB and AC are 400 mm² and 200 mm², respectively



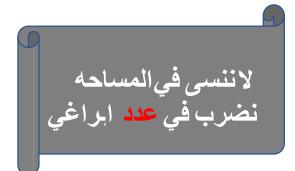
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Q1/If each of the three nails has a diameter of 4 mm and can withstand an average shear stress of 60 MPa , determine the maximum allowable force P that can be applied to the board



$$au = rac{V}{A}$$



$$V = \tau \times A$$

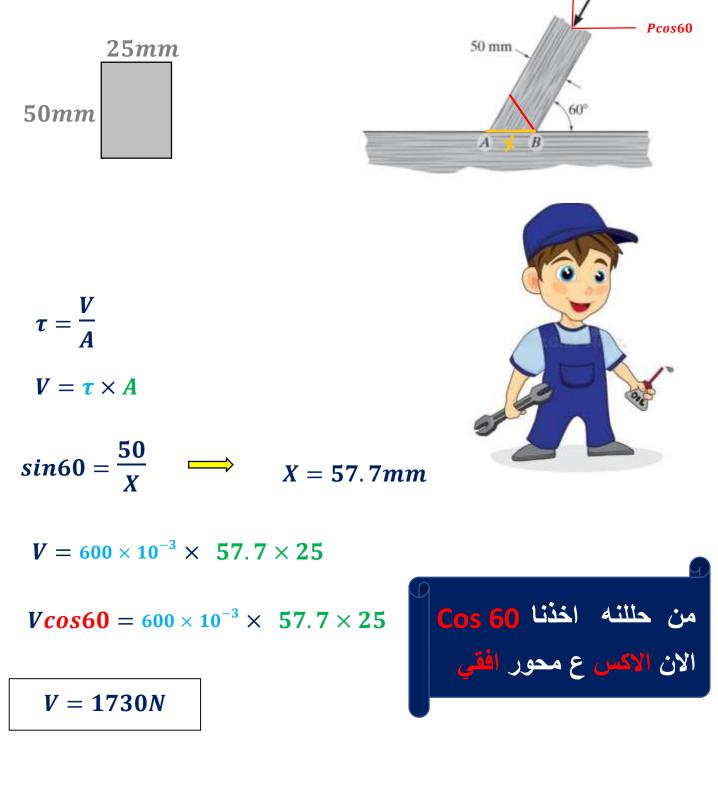
$$V = \tau \times \frac{\pi}{4} \times d^2 \times 3$$

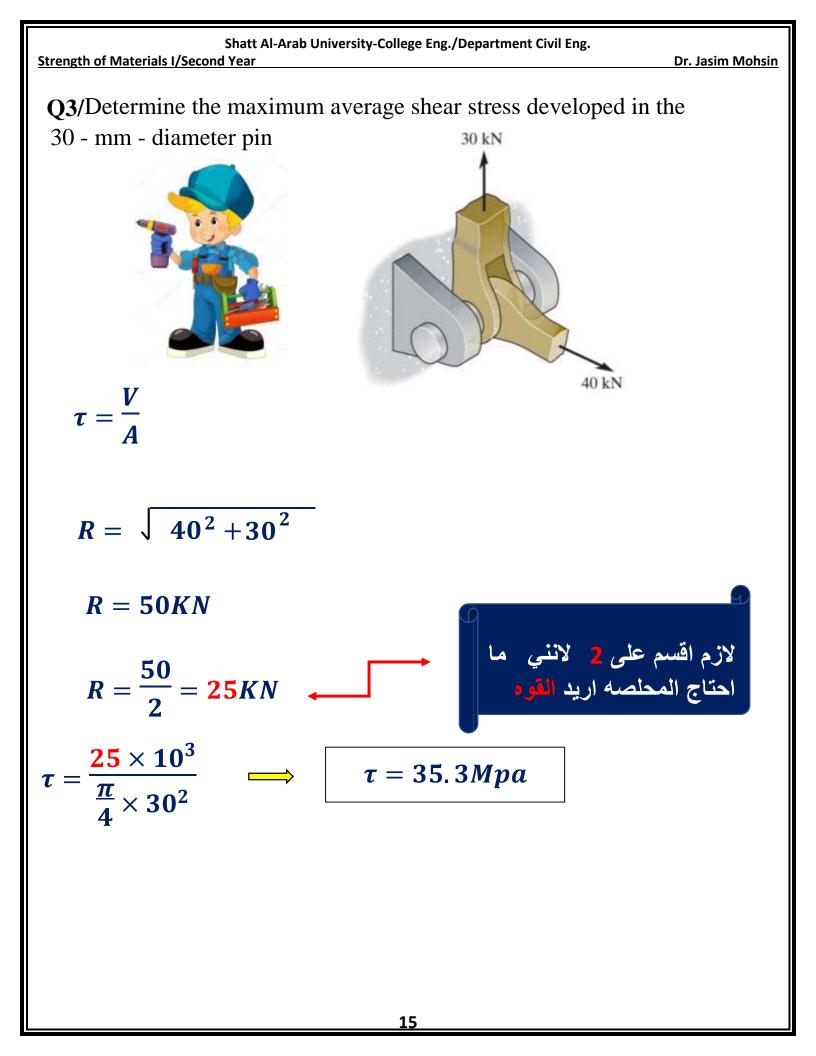
$$V = 60 \times \frac{\pi}{4} \times 4^2 \times 3$$

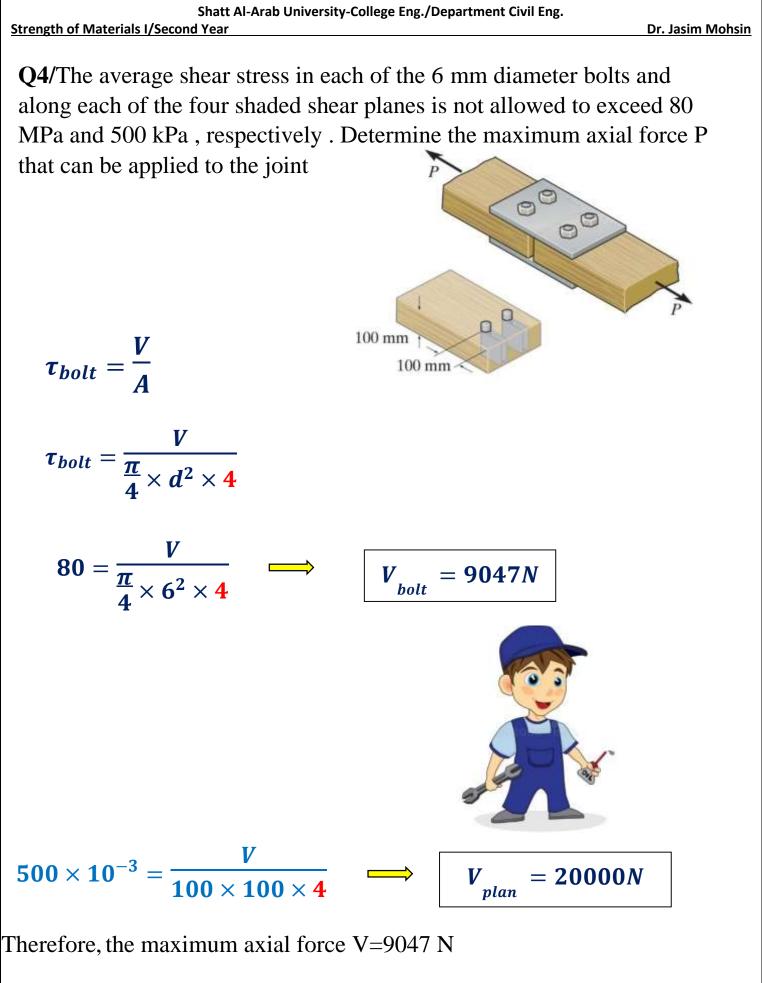
V = 2260N



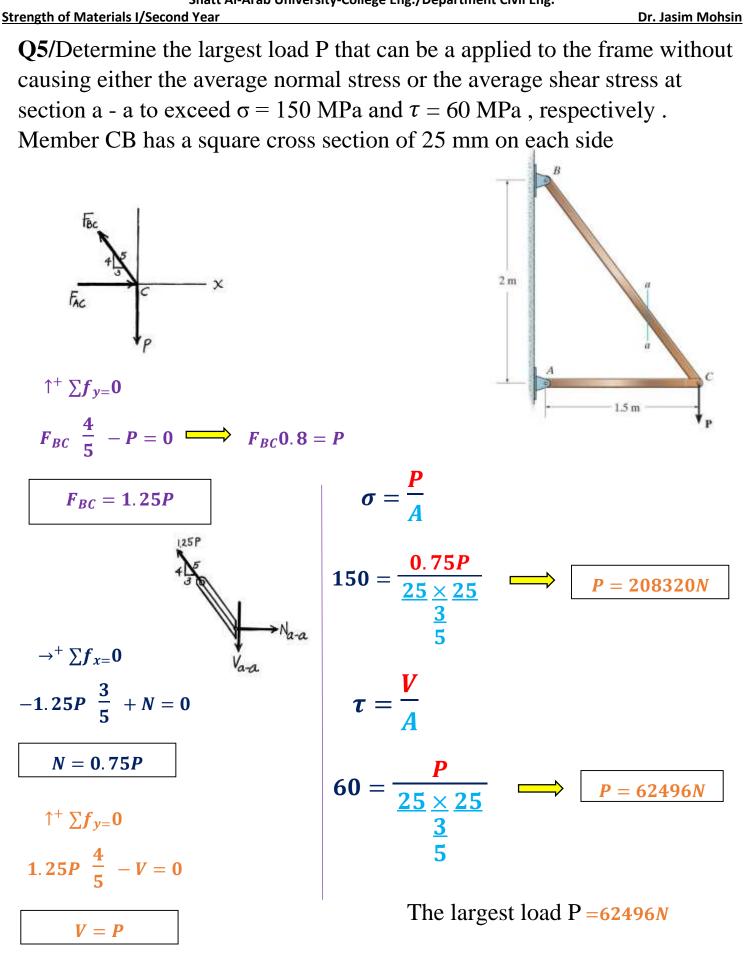
Q2/The strut is glued to the horizontal member at surface AB. If the strut has a thickness of 25 mm and the glue can withstand an average shear stress of 600 kPa, determine the maximum force P that can be applied to the strut Psin60 | P/P

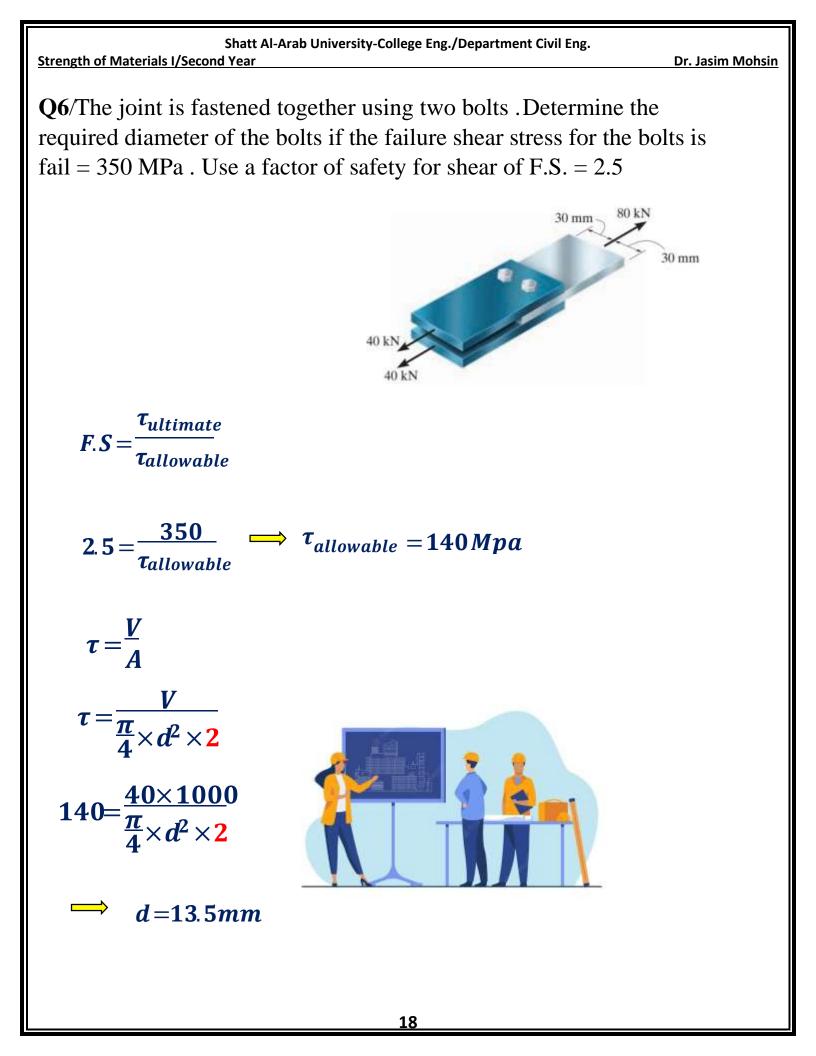


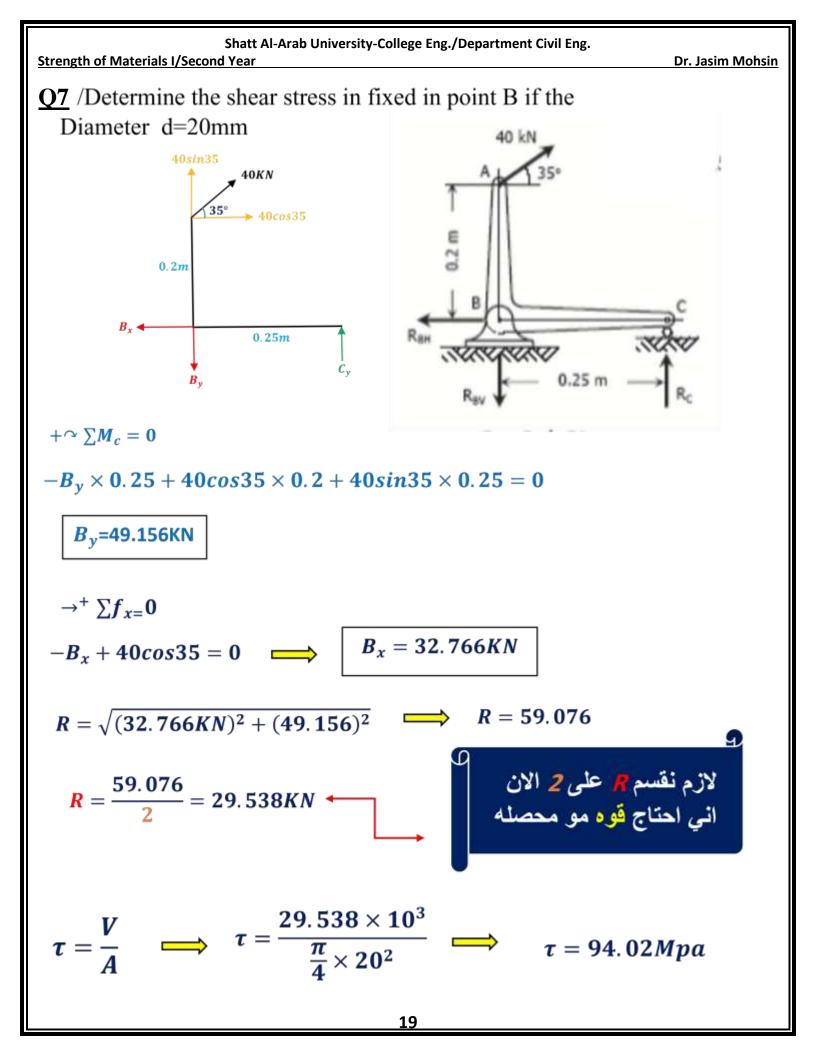


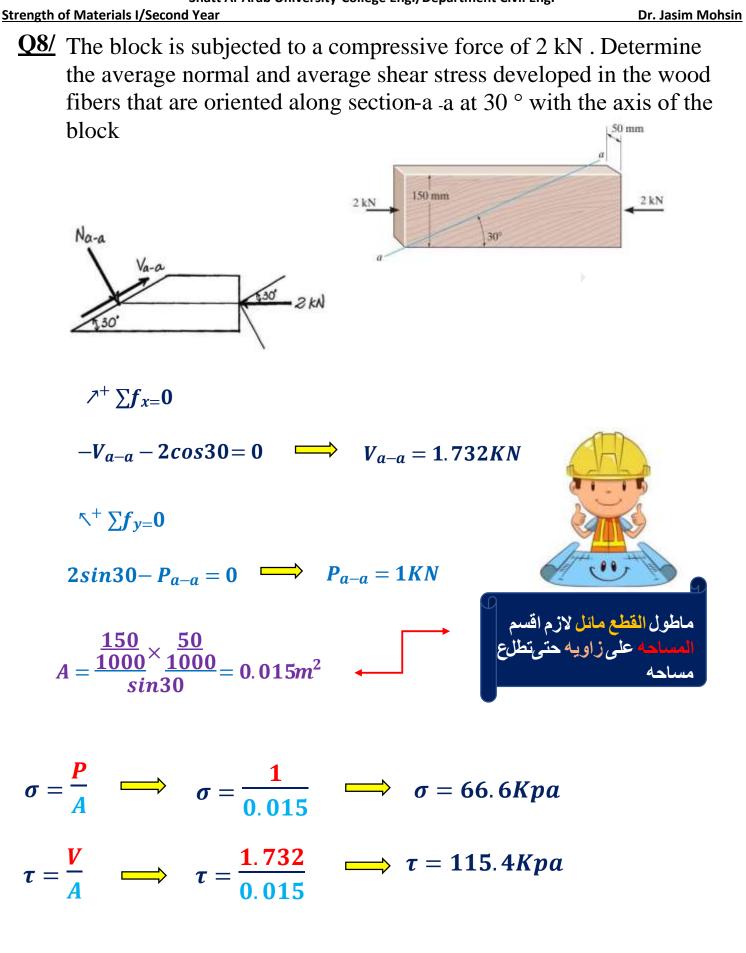


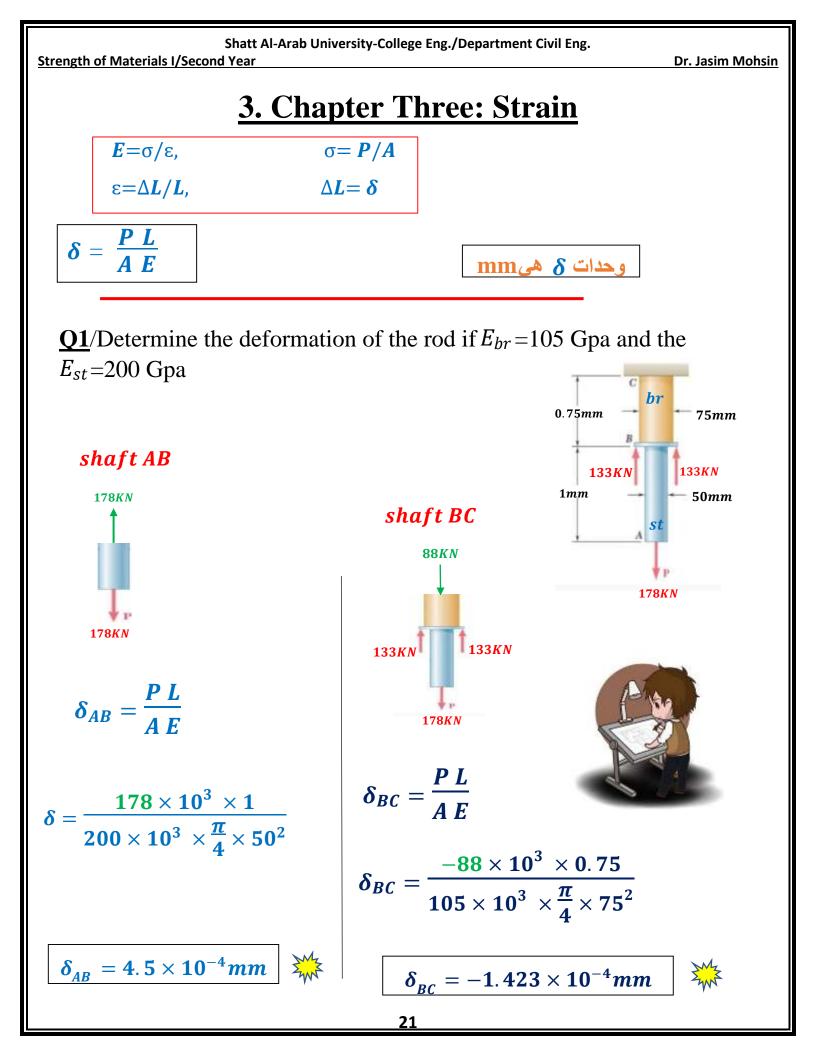


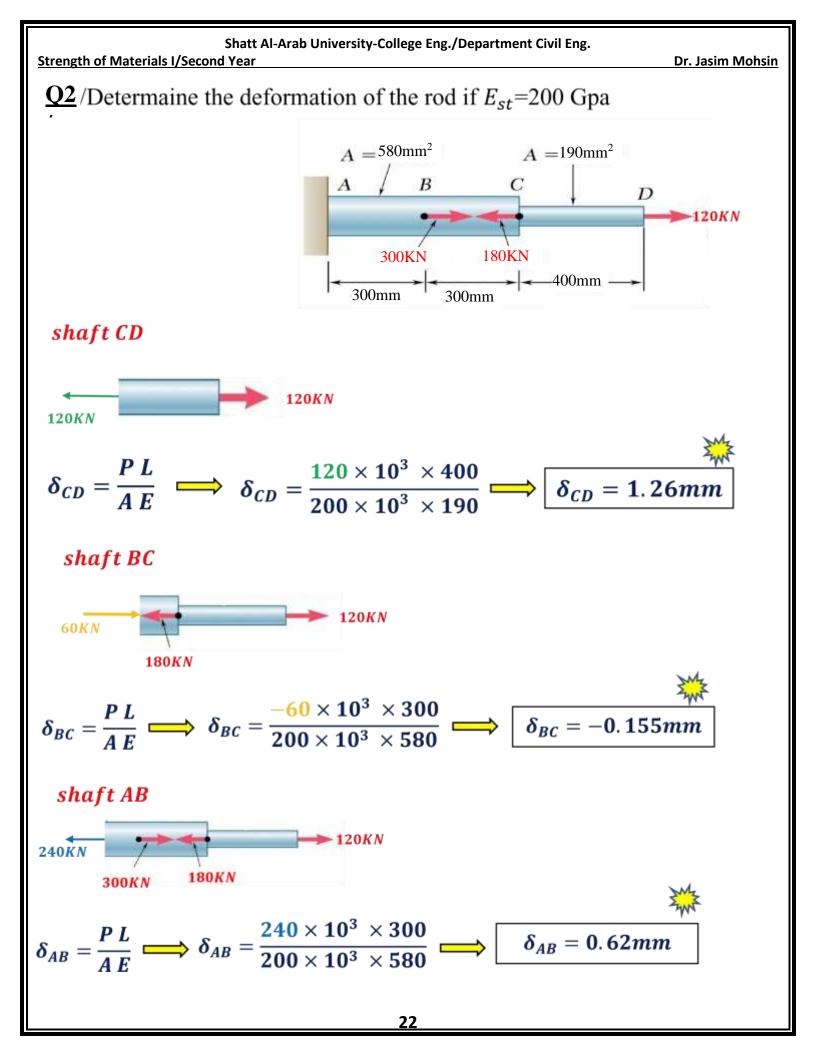






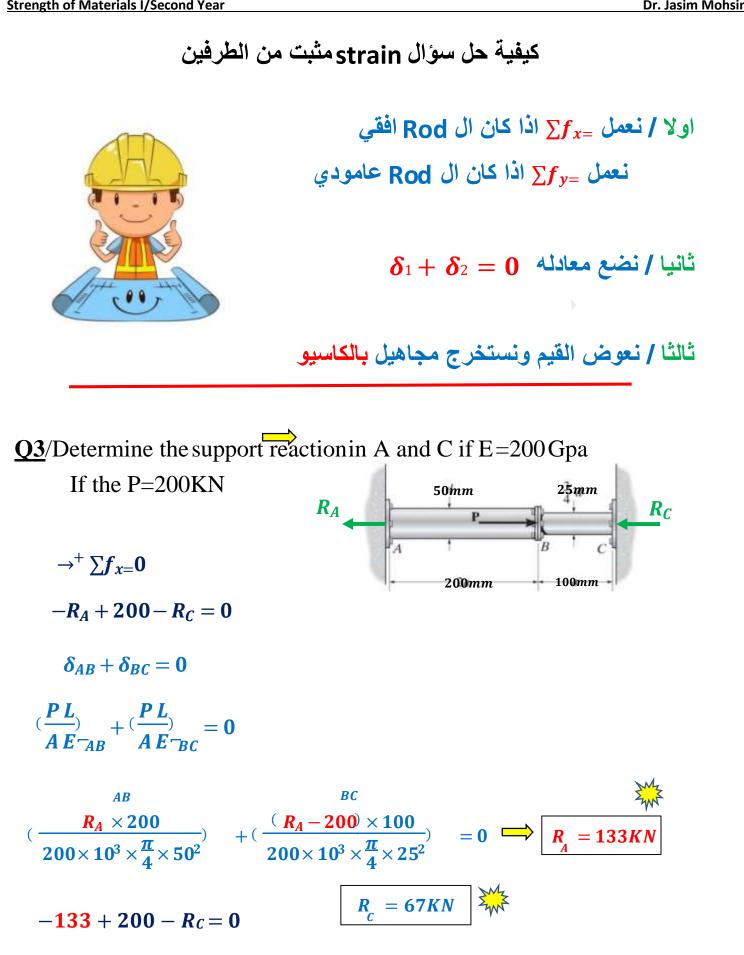




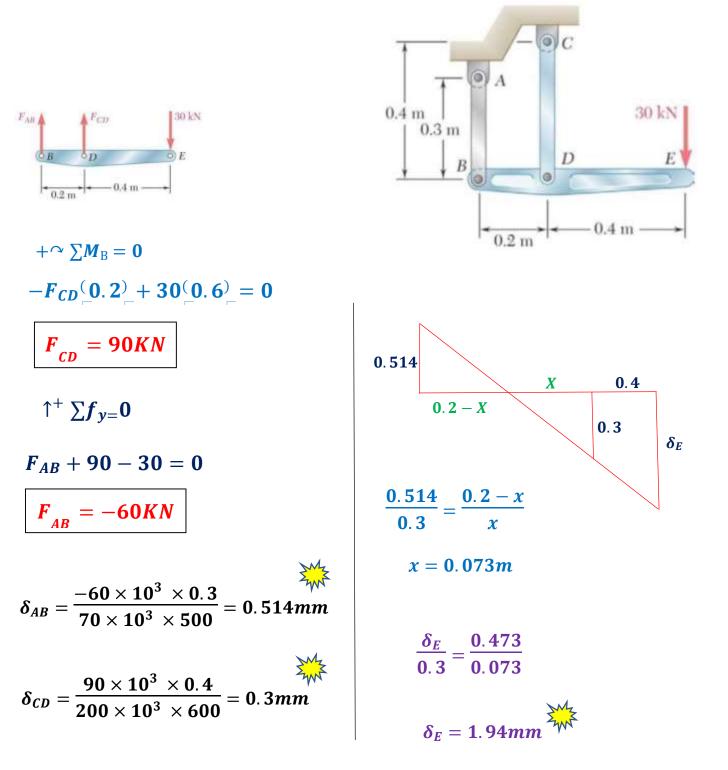


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<u>Q4/</u>The rigid bar BDE is supported by two links AB and CD . Link AB is made of aluminum (E = 70 GPa) and has a cross - sectional area of 500 mm²; link CD is made of steel (E = 200 GPa) and has a cross - sectional area of 600 mm². For the 30 - kN force shown, determine the deflection (a) of B. (b) of D. (c) of E.

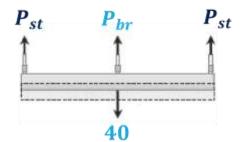


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<u>Q5/</u>The lower ends of the three bars in are at the same level before the uniform rigid block weighing 40 kips is attached. Each steel bar has a length of 3 ft, and area of $1.0 in^2$. and $E = 29 \times 10^6$ psl. For the bronze bar, the area is $1.5 in^2$. and $E = 12 \times 10^6$ " psi. Determine (a) the length of the bronze bar so that the load on each steel bar is twice the load on the bronze bar, and (b) the length of the bronze that will make the steel stress twice the bronze stress

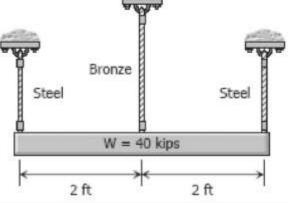
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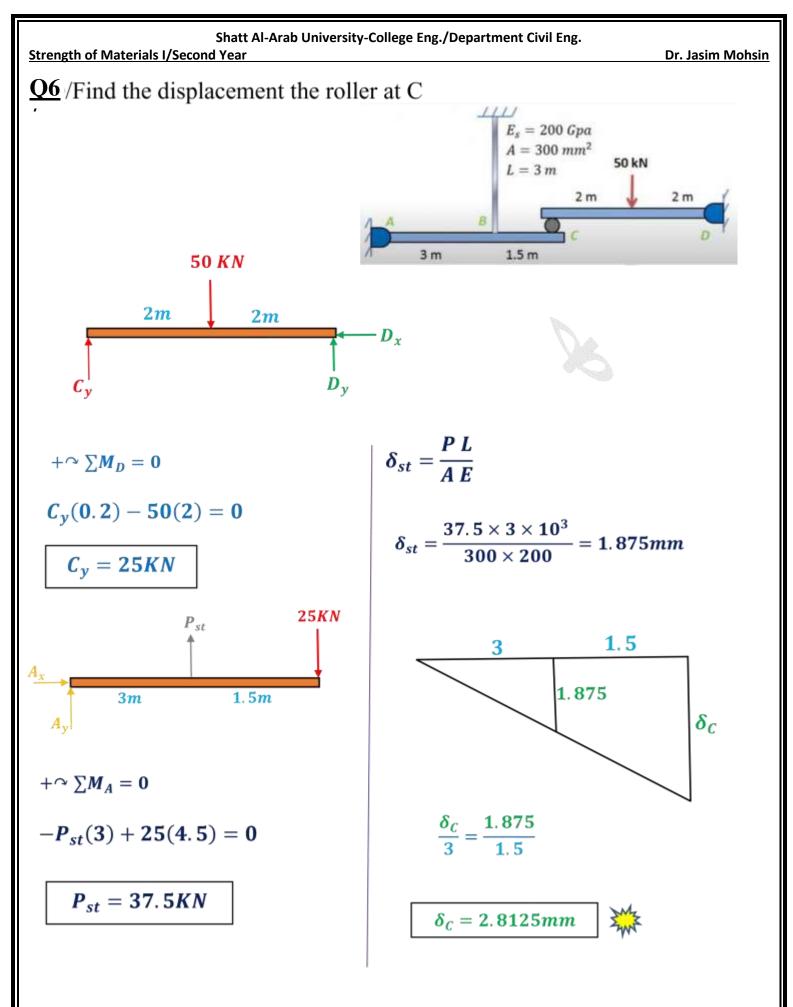
- $P_{st} = 2P_{br}$
- $\uparrow^+ \sum f_{y=0}$
- $2P_{st} + P_{br} = 40$
- $2(2P_{br}) + P_{br} = 40$ $P_{br} = 8Kip$
 - $P_{st} = 16Kip$

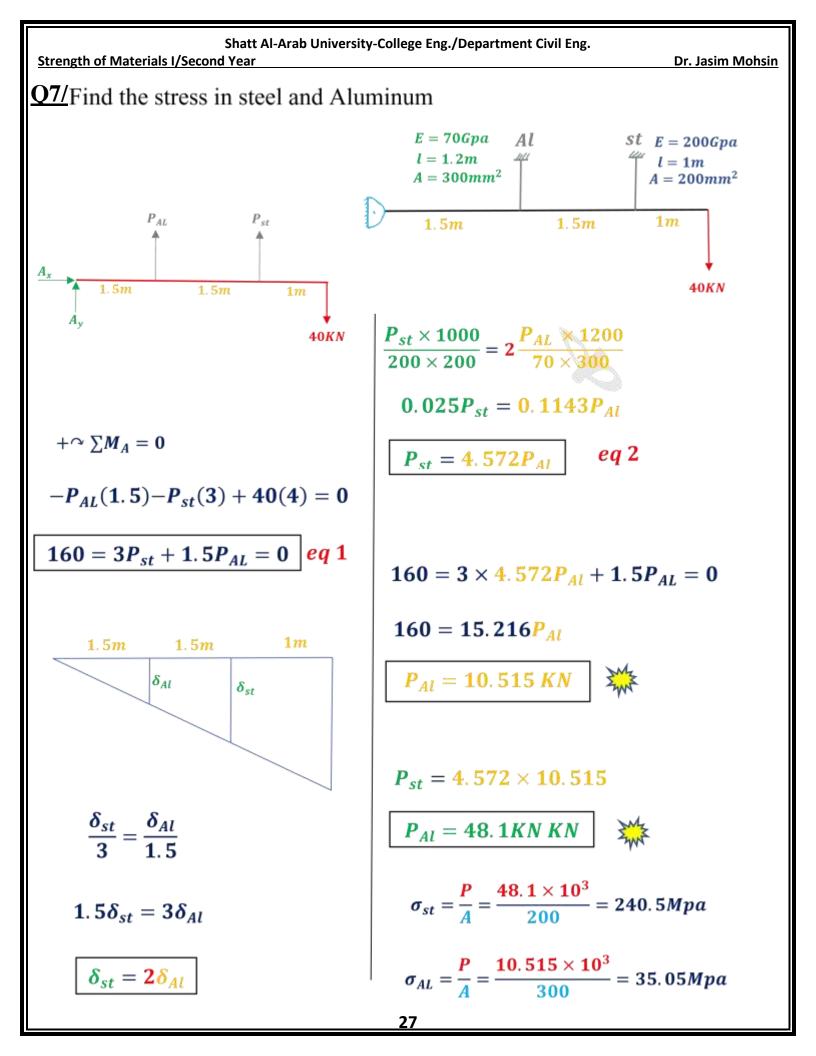
 $\delta_{br} = \delta_{st}$

 $\frac{8 \times 10^{3} \times l_{br}}{12 \times 10^{6} \times 1.5} = \frac{16 \times 10^{3} \times 3 \times 12}{29 \times 10^{6} \times 1.5}$ $l_{br} = 44.69 in$

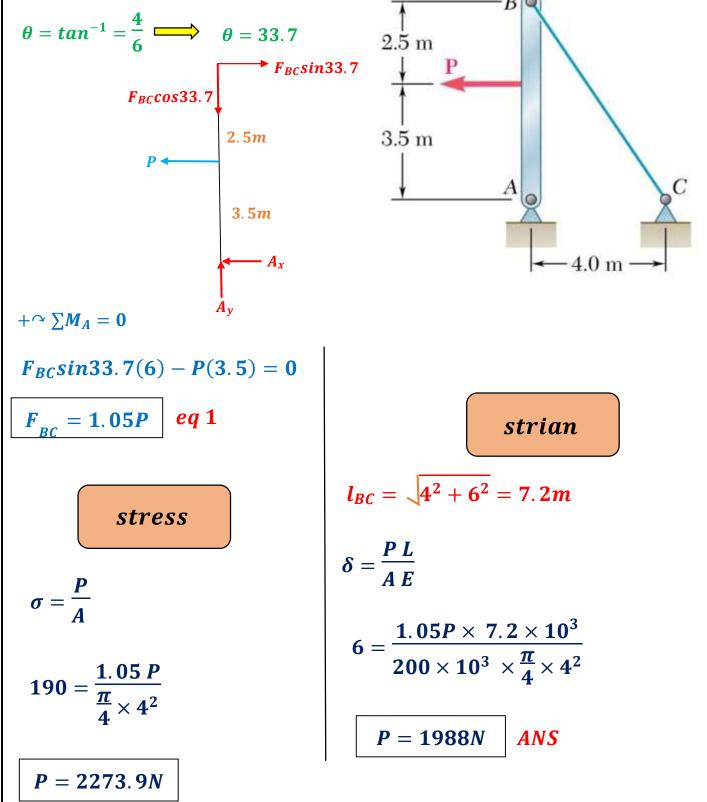


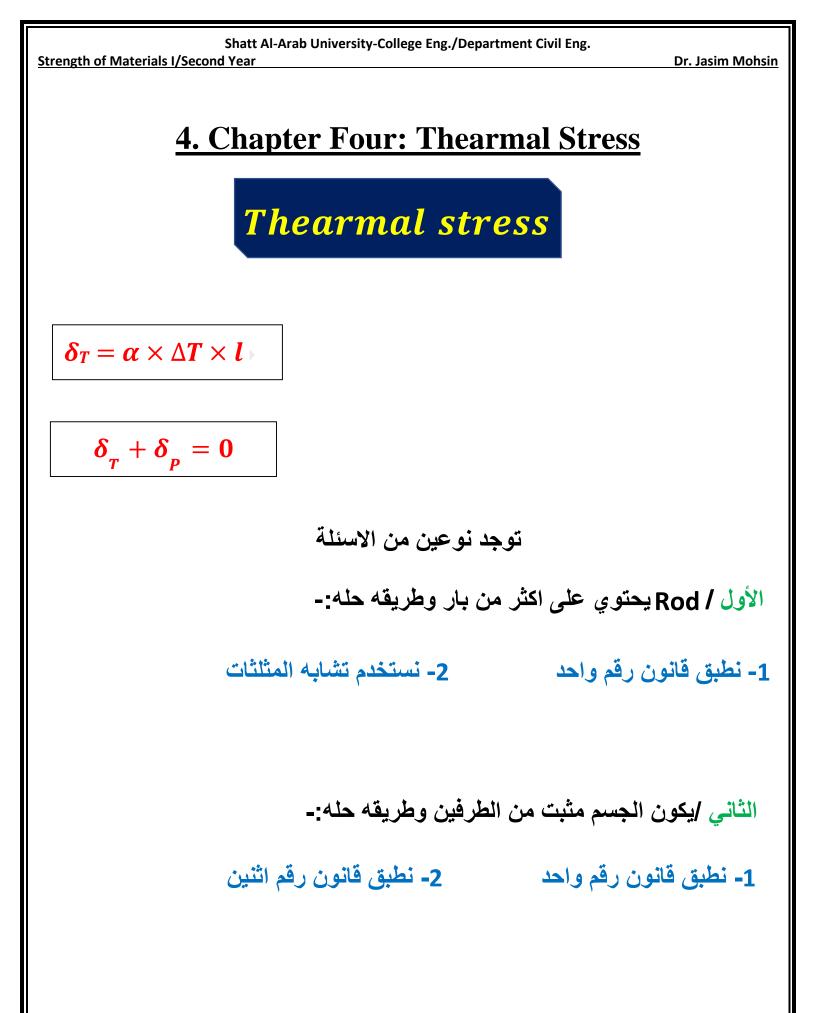
 $\sigma_{st} = 2\sigma_{br}$ $\uparrow^{+} \sum f_{y=0}$ $2P_{st} + P_{br} = 40$ $2(\sigma_{st} \times A_{st}) + (\sigma_{br} \times A_{br}) = 40$ $2(2\sigma_{br} \times A_{st}) + (\sigma_{br} \times A_{br}) = 40$ $4\sigma_{br} \times 1 + \sigma_{br} \times 1.5 = 40$ $\sigma_{br} = 7.27Ksi$ $\sigma_{st} = 14.54Ksi$ $\delta_{br} = \delta_{st}$ $\frac{7.27 \times 10^{3} \times l_{br}}{12 \times 10^{6}} = \frac{14.54 \times 10^{3} \times 3 \times 12}{29 \times 10^{6}}$ $l_{br} = 29.79 in$

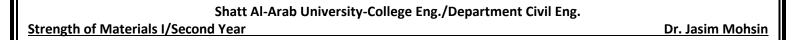




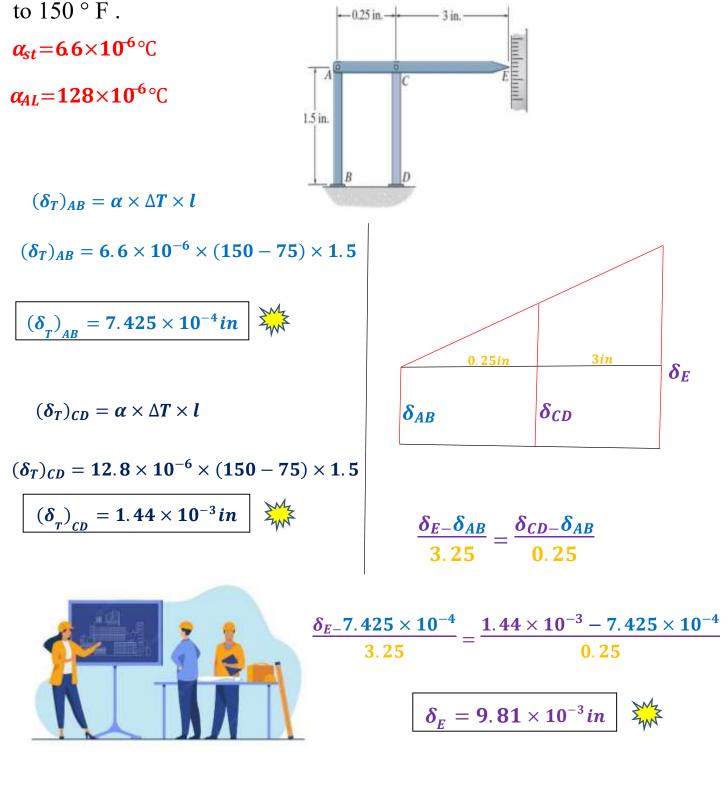
<u>**O8**</u>/The 4 mm - diameter cable BC is made of a steel with E = 200 GPa Knowing that the maximum stress in the cable must not exceed 190 MPa and that the elongation of the cable must not exceed 6 mm, find the maximum load P that can be applied as shown







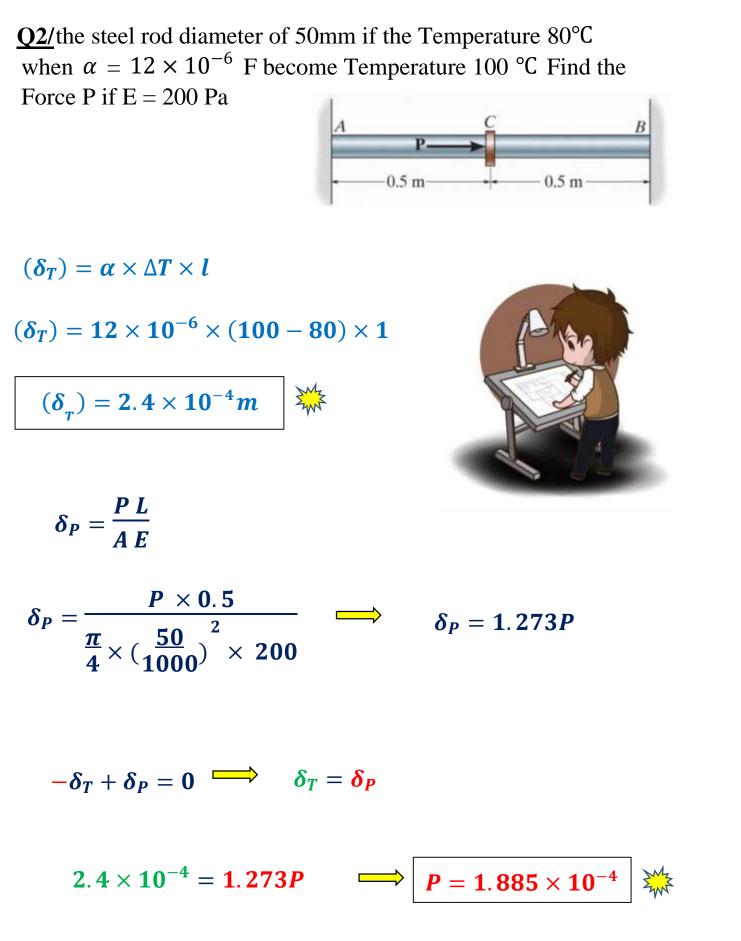
<u>Q1</u>/The device is used to measure a change in temperature. Bars AB and CD are made of steel and aluminum alloy, respectively. When the temperature is at 75 ° F, ACE is in the horizontal position. Determine the vertical displacement of the pointer at E when the temperature rises to $150 \circ F$.

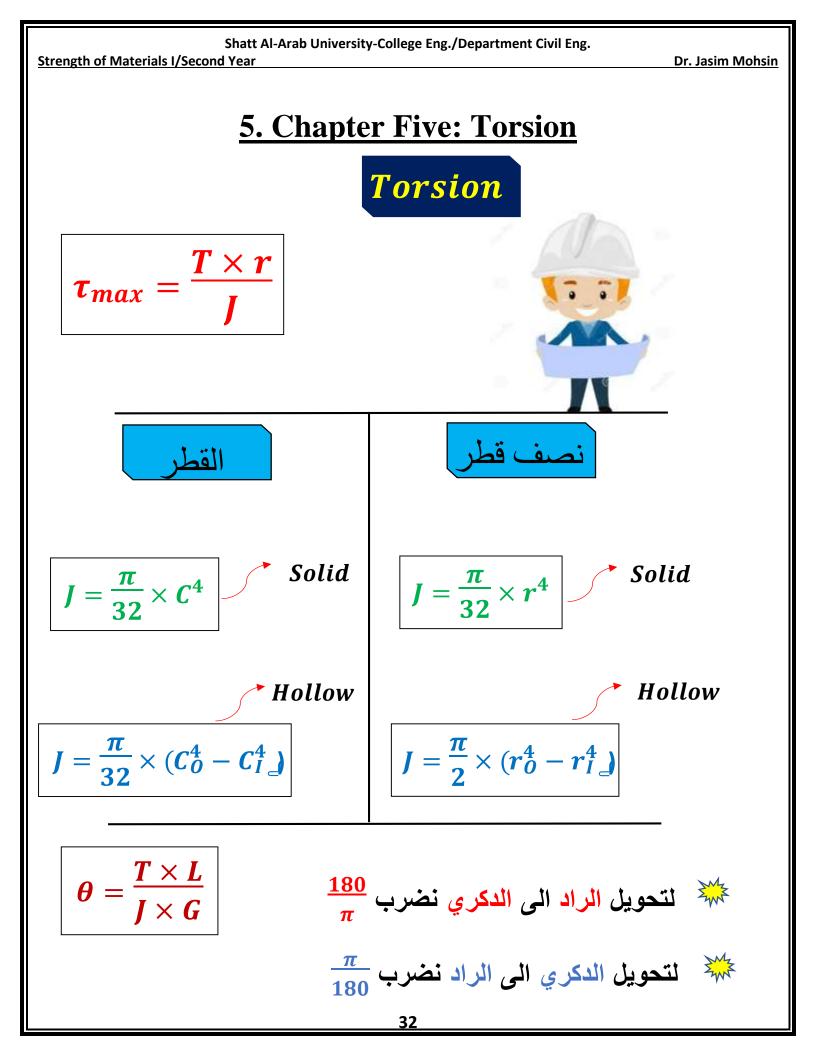


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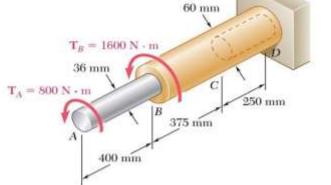


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<u>Q1/</u>The aluminum rod AB (G = 27 GPa) is bonded to the brass rod BD (C = 39 GPa). Knowing that portion CD of the brass rod is hollow and has an inner diameter of 40 mm , determine the angle of twist at A



 $\boldsymbol{\theta}_A = \boldsymbol{\theta}_{AB} + \boldsymbol{\theta}_{BC} + \boldsymbol{\theta}_{CD}$

$$\theta_{AB} = \frac{T \times L}{J \times G} \implies \theta_{AB} = \frac{800 \times 10^3 \times 400}{\frac{\pi}{2} \times 18^4 \times 27 \times 10^3} \implies \theta_{AB} = 0.0719 \, rad$$

$$\theta_{BC} = \frac{T \times L}{J \times G} \implies \theta_{BC} = \frac{2400 \times 10^3 \times 375}{\frac{\pi}{2} \times 30^4 \times 39 \times 10^3} \implies \theta_{BC} = 0.018 \, rad$$

 $\theta_{CD} = \frac{T \times L}{J \times G} \implies \theta_{BC} = \frac{2400 \times 10^3 \times 250}{\frac{\pi}{2} \times (30^4 - 20^4) \times 39 \times 10^3} \implies \theta_{CD} = 0.015 rad$ $\theta_A = \theta_{AB} + \theta_{BC} + \theta_{CD}$ $\theta_A = 0.0719 + 0.018 + 0.015 \implies \theta_A = 0.1049 rad$

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<u>Q2</u> compound shaft consisting of a steel segment and an aluminum segment is acted upon by two torques as shown in the figure . Determine the maximum permissible value of T subject to the following conditions : $\tau_{st} = 83$ MPa , $\tau_{al} = 55$ MPa , and the angle of rotation of the free end is limited to 6°. For steel , G = 83 GPa and for aluminum , G = 28 GPa

Part1

$$\tau_{al}=\frac{T\times r}{J}$$

 $\tau_{al} = \frac{T \times 20}{\frac{\pi}{2} \times 20^4}$

 $\tau_{al} = 691150.38 N.mm$

$$\tau_{al} = 691.15 \, \text{N.m}$$

Pa
Steel 2T Aluminum 1
40 mm Ø 40 mm Ø 600 mm
Part2

$$\tau_{st} = \frac{T \times r}{J}$$

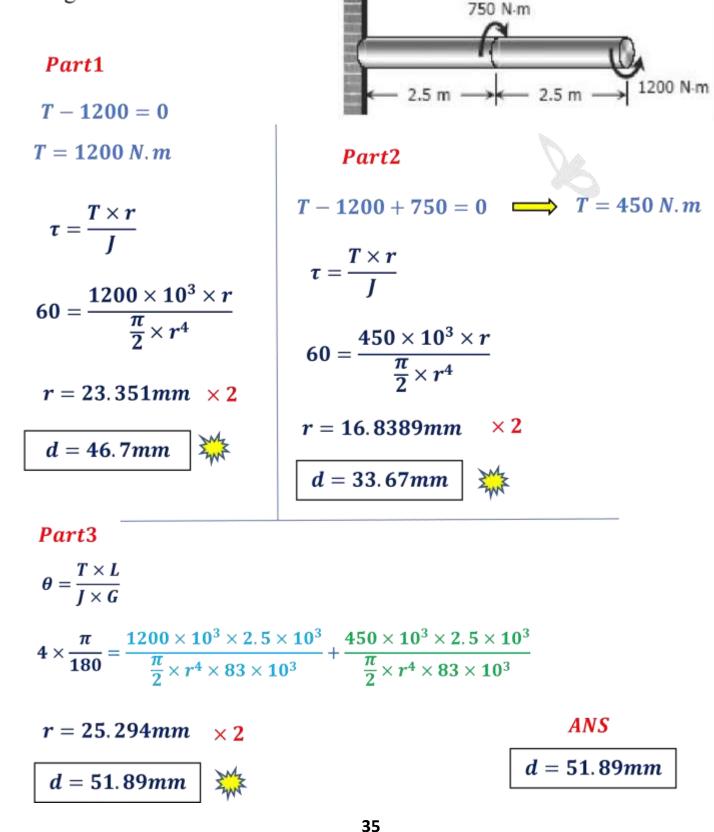
 $\tau_{st} = \frac{3T \times 25}{\frac{\pi}{2} \times 25^4}$
 $\tau_{st} = 679042.16 N.mm$

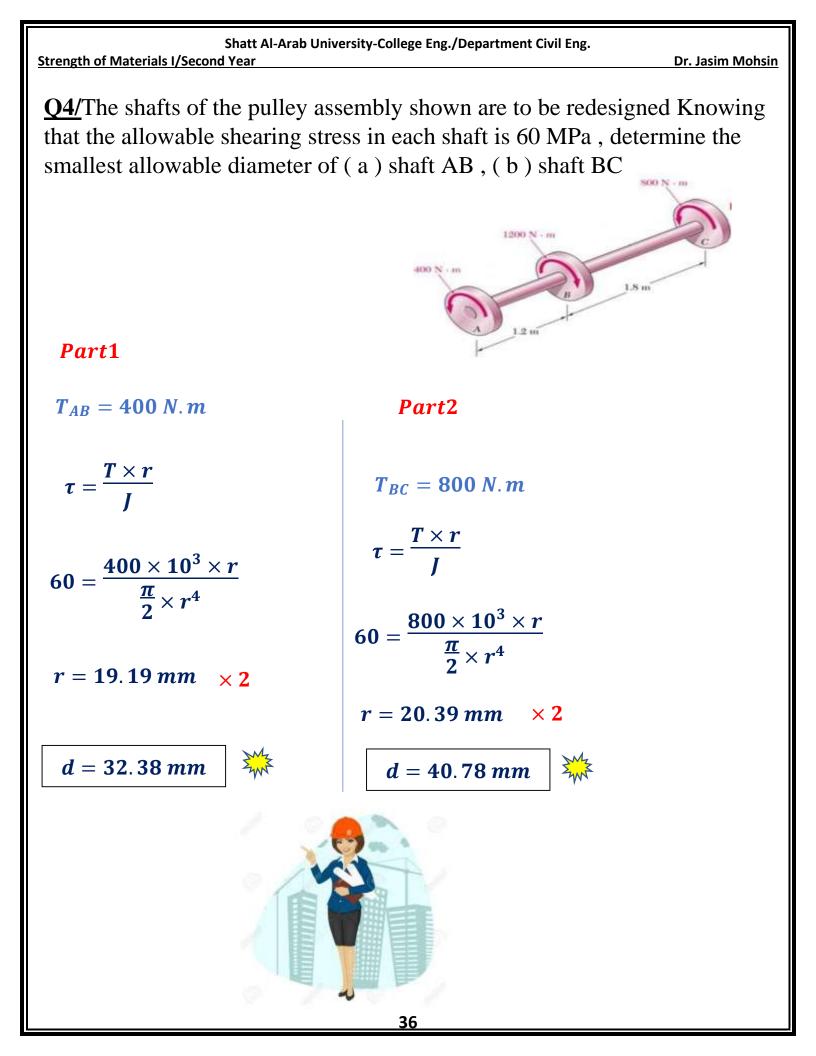
Part3

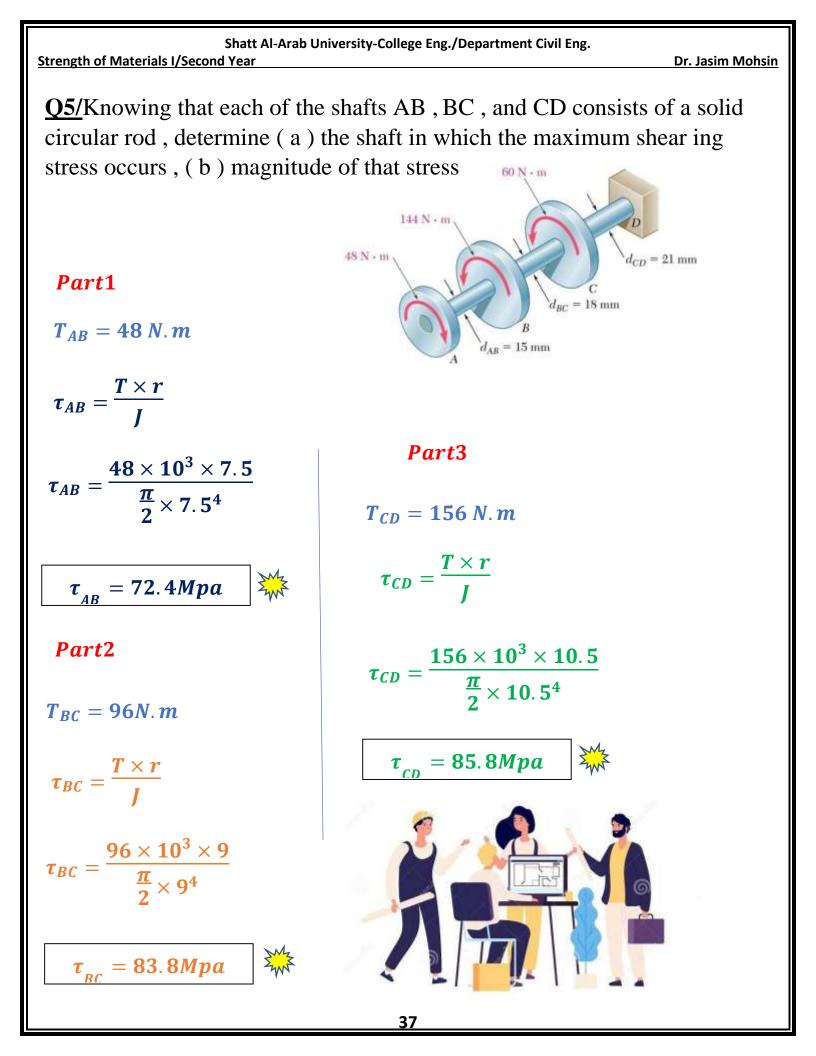
 $\theta = \theta_{al} + \theta_{st}$ $6 \times \frac{\pi}{180} = \frac{T \times 600}{\frac{\pi}{2} \times 20^4 \times 28 \times 10^3} + \frac{3T \times 900}{\frac{\pi}{2} \times 25^4 \times 83 \times 10^3}$ $\theta = 757316.32 N.mm$ $\theta = 757.32 N.m$ $\tau_{st} = 679.04 N.m$

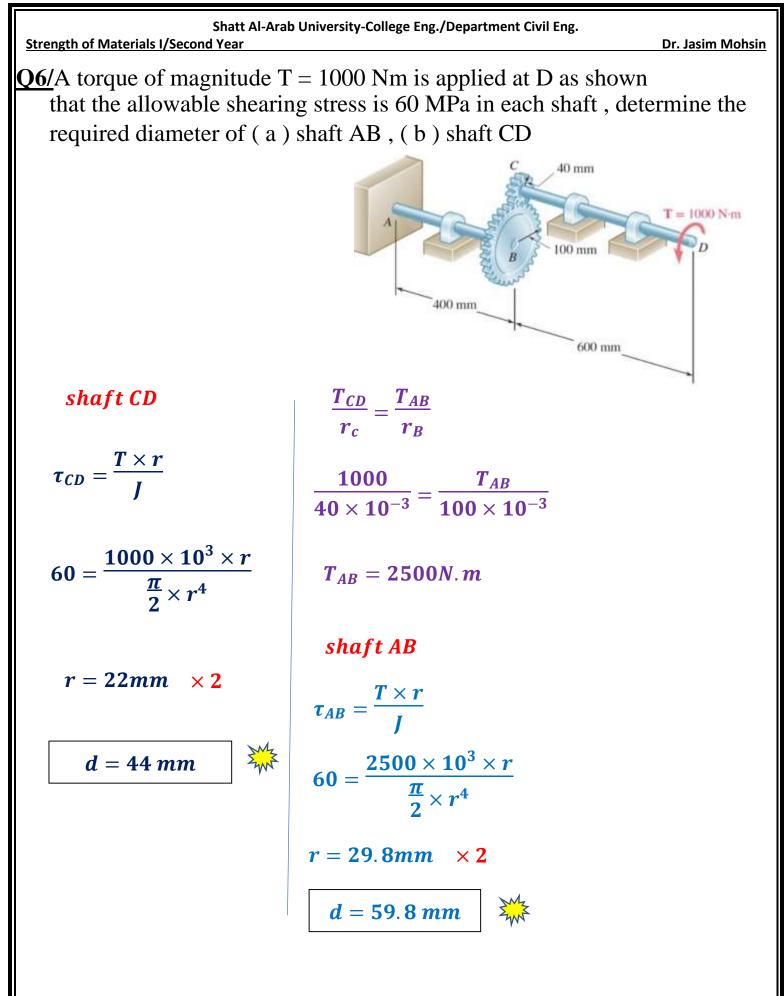
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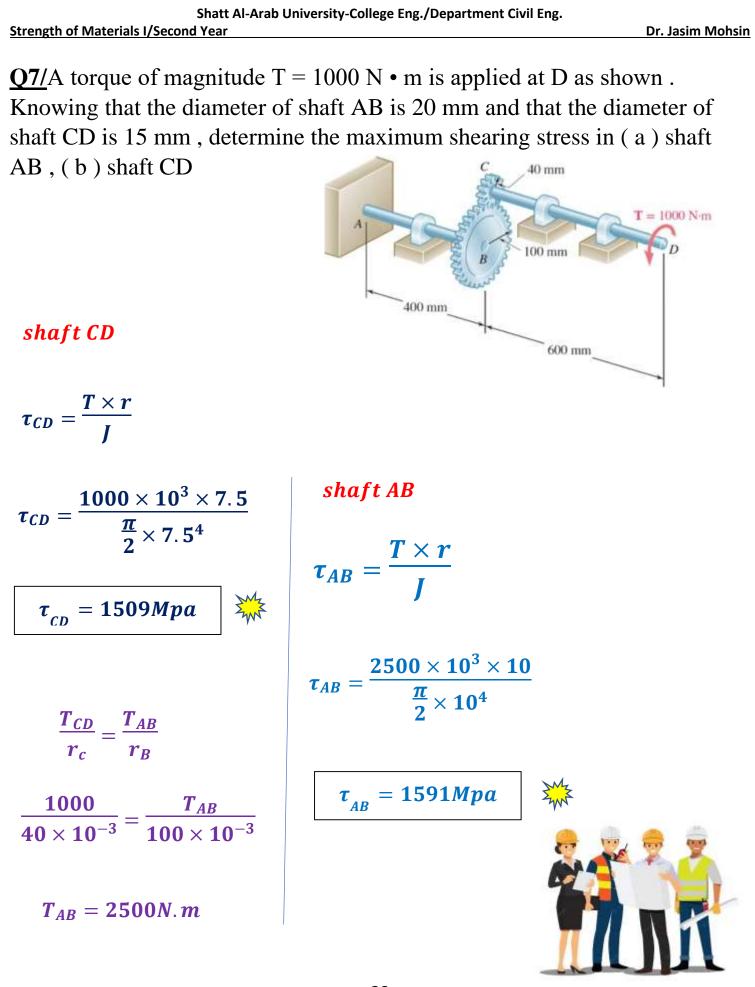
Q3A solid steel shaft is loaded as shown in Fig . P - 322 . Using G = 83 GPa , determine the required diameter of the shaft if the shearing stress is limited to 60 MPa and the angle of rotation at the free end is not to exceed 4 deg













Q8/The two solid shafts are connected by gears as shown and are made of a steel for which the allowable shearing stress is alle 7.0 ksi. This gear set is designed to lock - in the torque applied at C and F, respectively, so that the gear set is in static equilibrium. Knowing the diameters of the two shafts are, respectively, $d_c=1.6$ in . and $d_F = 1.25$ in . , determine the largest torque T_c that can be applied at C.

$$\tau = \frac{I_C \times I}{J}$$
$$T_C \times 0.8$$

$$7 = \frac{\pi}{2} \times 0.8^4$$

$$T_c = 5.63 Kip.in$$

$$\tau = \frac{T_F \times r}{J}$$

$$7 = \frac{T_F \times 0.625}{\frac{\pi}{2} \times 0.625^4}$$

$$T_F = 2.68 \, Kip \, . in$$

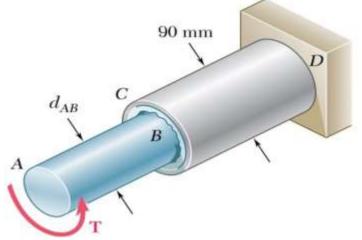
2.5 in

$$\frac{T_C}{4} = \frac{T_F}{2.5}$$

$$\frac{T_c}{4} = \frac{2.68}{2.5}$$

$$T_c = 4.288 \, Kip \, . \, in$$

<u>**Q9/**</u>The solid rod AB has a diameter dB 60 mm and is made of a steel for which the allowable shearing stress is 85 MPa. The pipe CD, which has an outer diameter of 90 mm and a wall thickness of 6 mm, is made of an aluminum for which the allowable shearing stress is 54 MPa. Determine the largest torque T that can be applied at A



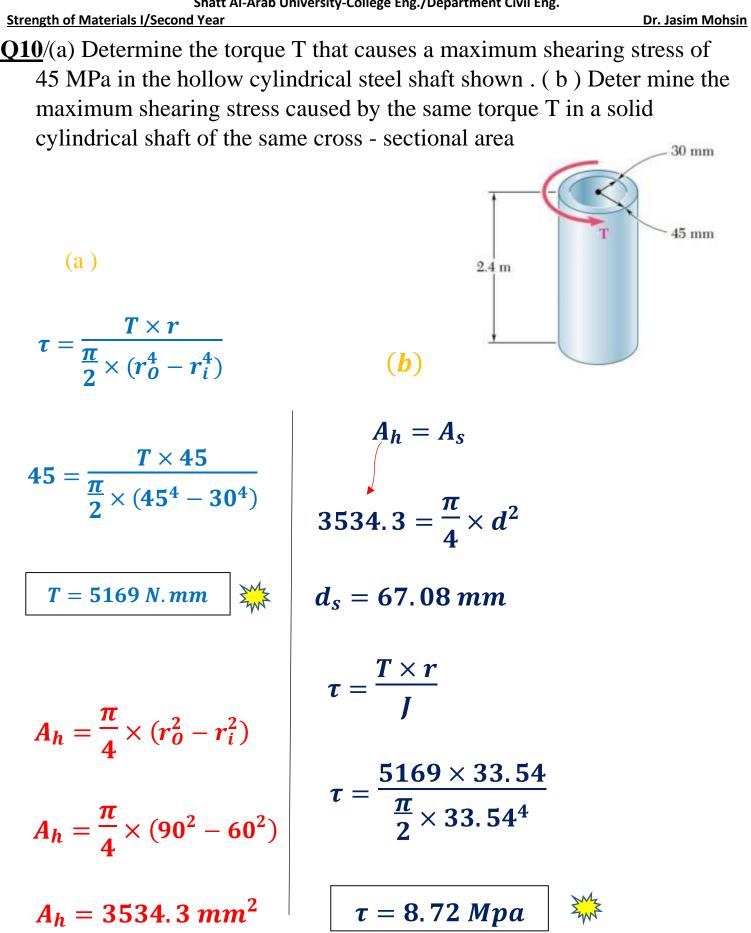
shaft AB

$$\tau = \frac{T \times r}{J}$$

$$85 = \frac{T \times 30}{\frac{\pi}{2} \times 30^{4}}$$

$$T = 3604977 N.mm$$

 $d_{i} = d_{o} - 2 \times t$ $d_{i} = 90 - 2 \times 6 = 78mm$ $\tau = \frac{T \times r}{\frac{\pi}{2} \times (r_{0}^{4} - r_{i}^{4})}$ $54 = \frac{T \times 45}{\frac{\pi}{2} \times (45^{4} - 39^{4})}$ T = 3368762 N.mm MS T = 3368762 N.mm

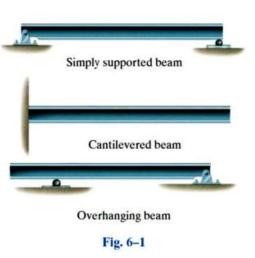


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6. Chapter Sex: Draw SFD and BMD

1. Introduction:

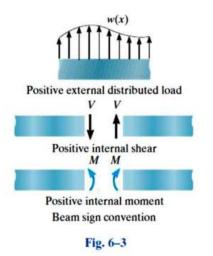
Members that are slender and support loadings that are applied perpendicular to their longitudinal axis are called beams. In general, beams are long, straight bars having a constant cross-sectional area. Often they are classified as to how they are supported. For example, a simply supported beam is pinned at one end and roller supported at the other, Fig. 6–1, a cantilevered beam is fixed at one end and free at the other, and an overhanging beam has one



or both of its ends freely extended over the supports. Beams are considered among the most important of all structural elements. They are used to support the floor of a building, the deck of a bridge, or the wing of an aircraft. Also, the axle of an automobile, the boom of a crane, even many of the bones of the body act as beams.

Beam Sign Convention. Before presenting a method for determining the shear and moment as functions of x and later plotting these functions (shear and moment diagrams), it is first necessary to establish a sign convention so as to define "positive" and "negative"

values for V and M . Although the choice of a sign convention is arbitrary, here we will use the one often used in engineering practice and shown in Fig. 6-3. The positive directions are as follows: the distributed load acts upward on the beam; the internal shear force causes a clockwise rotation of the beam segment on which it acts; and the internal moment causes compression in the top



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2. Method of Drawing of Shear Force and Bending Moment

2.1 Section Method

Procedure for Analysis

The shear and moment diagrams for a beam can be constructed using the following procedure.

a) Support Reactions.

• Determine all the reactive forces and couple moments acting on the beam, and resolve all the forces into components acting perpendicular and parallel to the beam's axis.

b) Shear and Moment Functions.

• Specify separate coordinates x having an origin at the beam's left end and extending to regions of the beam between concentrated forces and/or couple moments, or where there is no discontinuity of distributed loading.

• Section the beam at each distance x, and draw the free-body diagram of one of the segments. Be sure V and M are shown acting in their positive sense, in accordance with the sign convention given in Fig. 6–3.

• The shear is obtained by summing forces perpendicular to the beam's axis.

• To eliminate V, the moment is obtained directly by summing moments about the sectioned end of the segment.

c) Shear and Moment Diagrams.

• Plot the shear diagram (V versus x) and the moment diagram (M versus x). If numerical values of the functions describing V and M are positive, the values are plotted above the x axis, whereas negative values are plotted below the axis.

• Generally it is convenient to show the shear and moment diagrams below the free-body diagram of the beam.

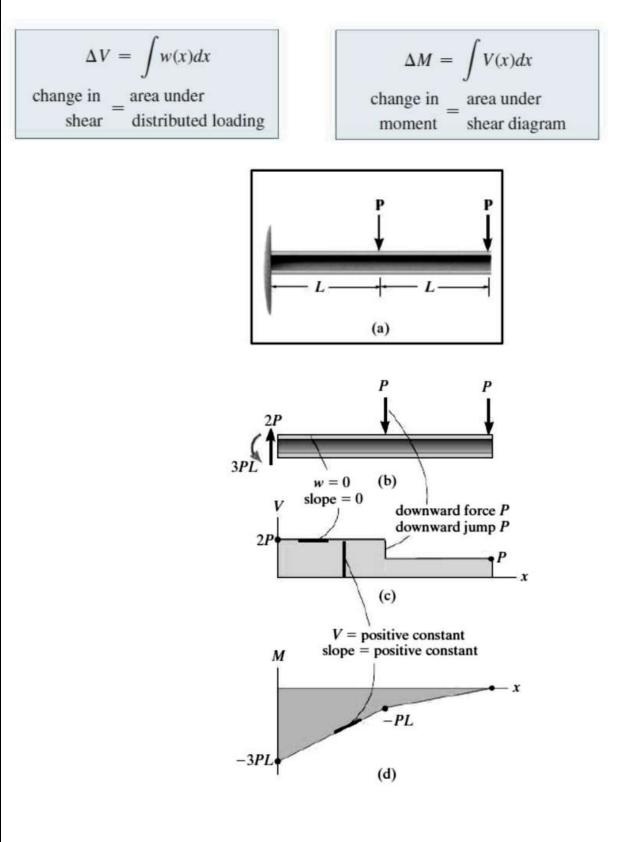
2.2 Graphical Method for Constructing Shear and moment diagrams

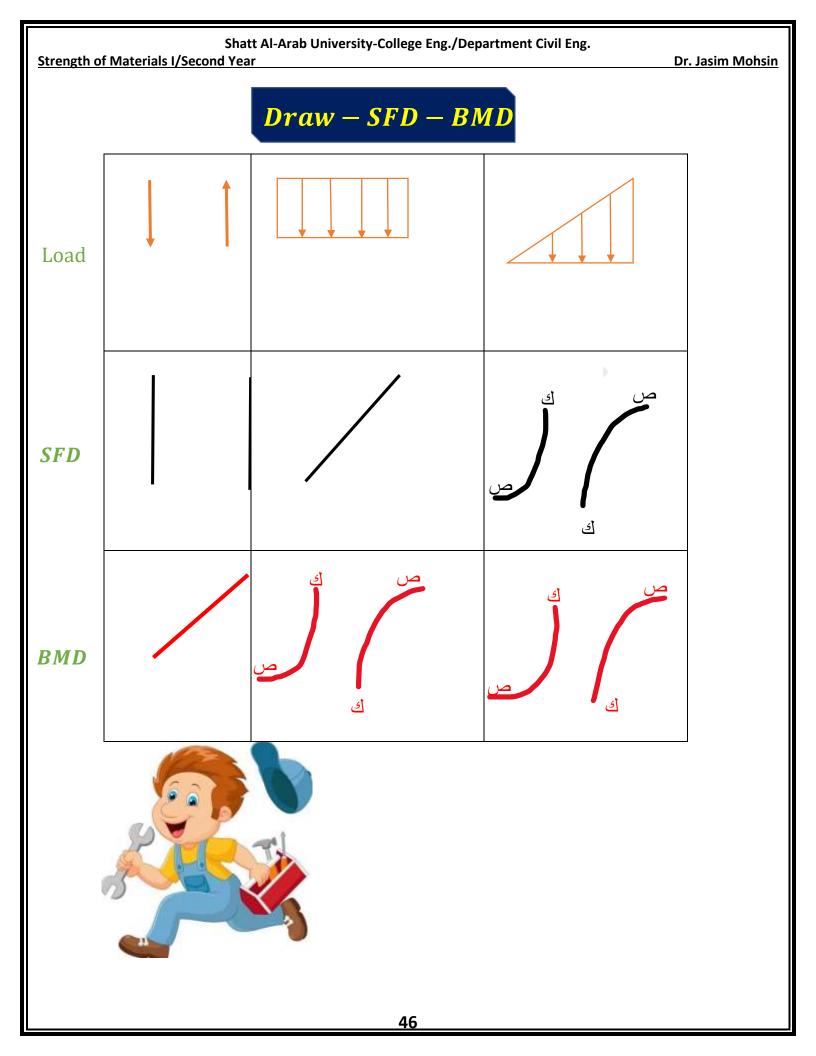
In cases where a beam is subjected to several different loadings, determining V and M as functions of x and then plotting these equations can become quite tedious. In this section a simpler method for constructing the shear and moment diagrams is discussed—a method based on two differential relations, one that exists between distributed load and shear, and the other between shear and moment.

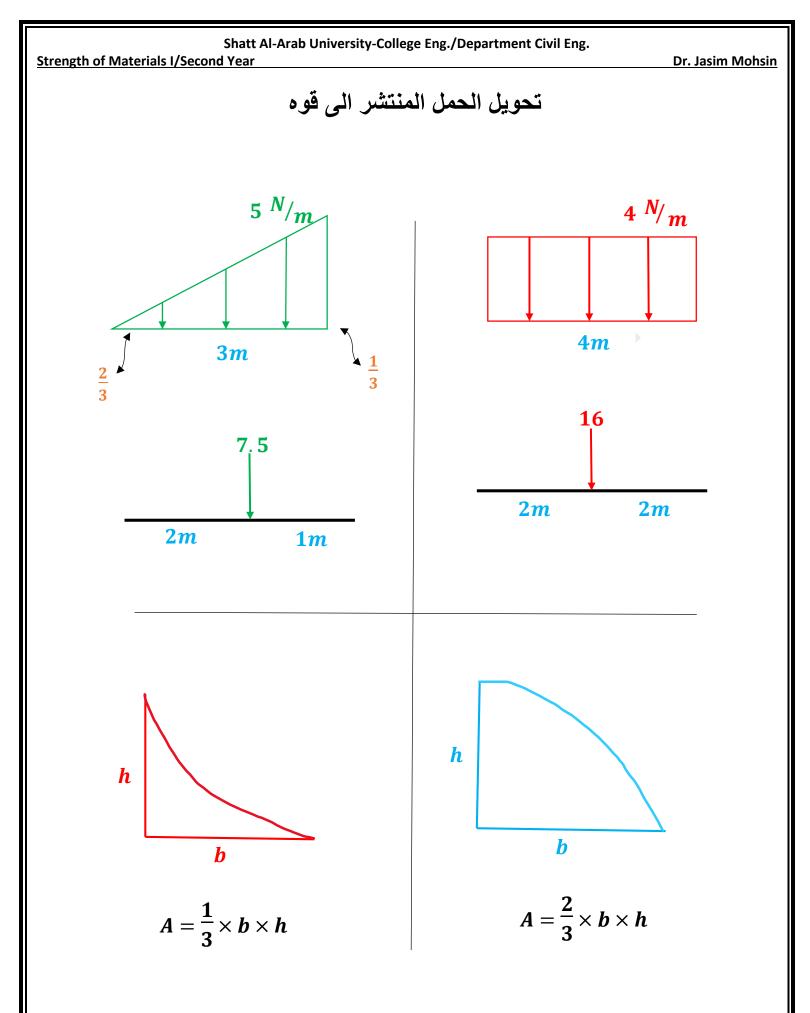
$\frac{dV}{dx} = w(x)$	$\frac{dM}{dx} = V(x)$
slope ofdistributedshear diagram=load intensityat each pointat each point	slope of shear moment diagram = at each at each point point

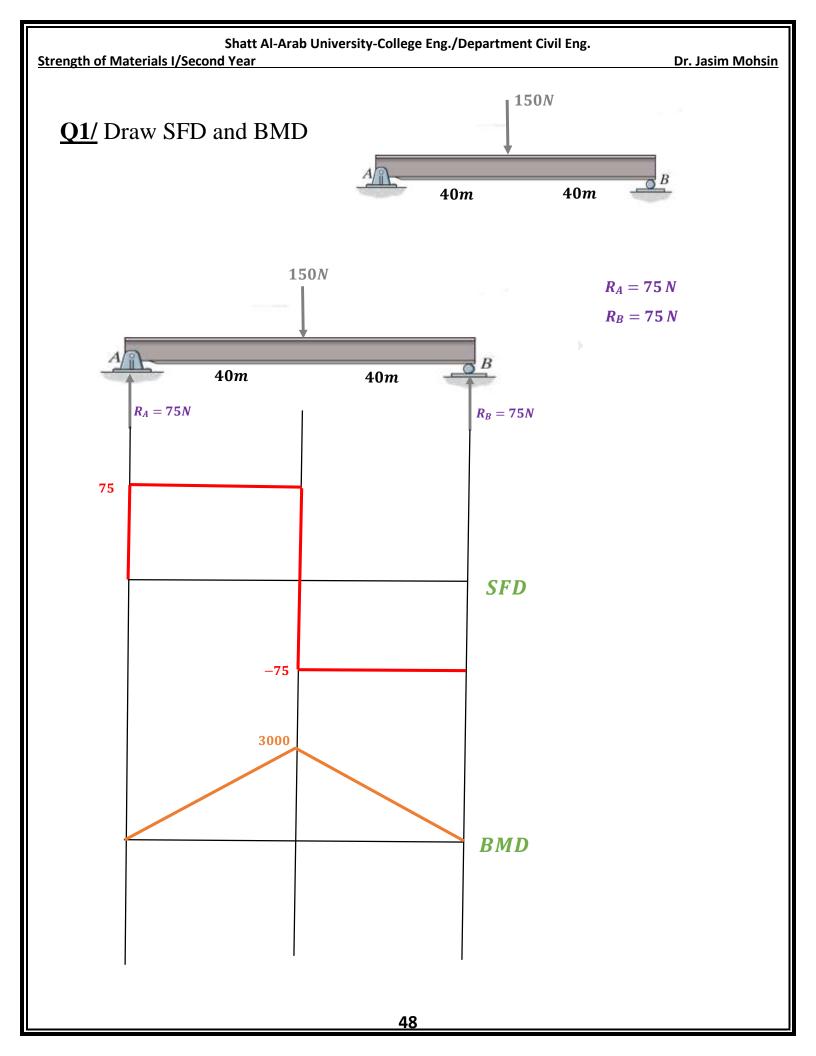
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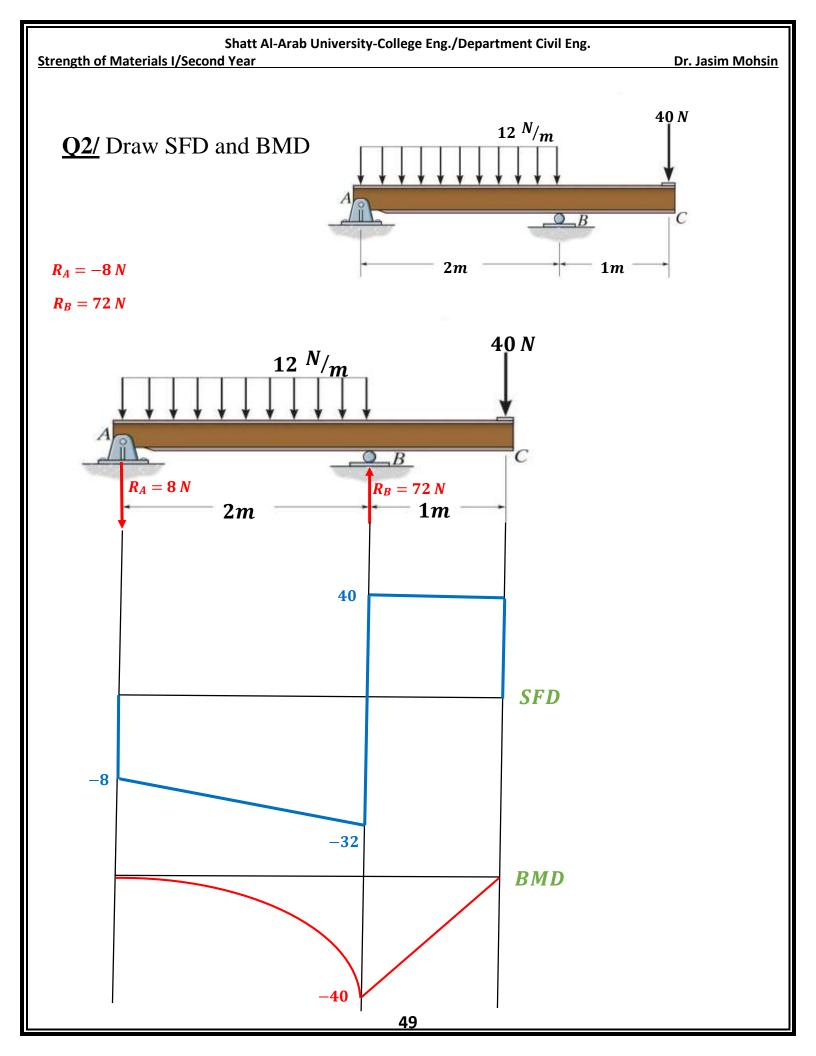
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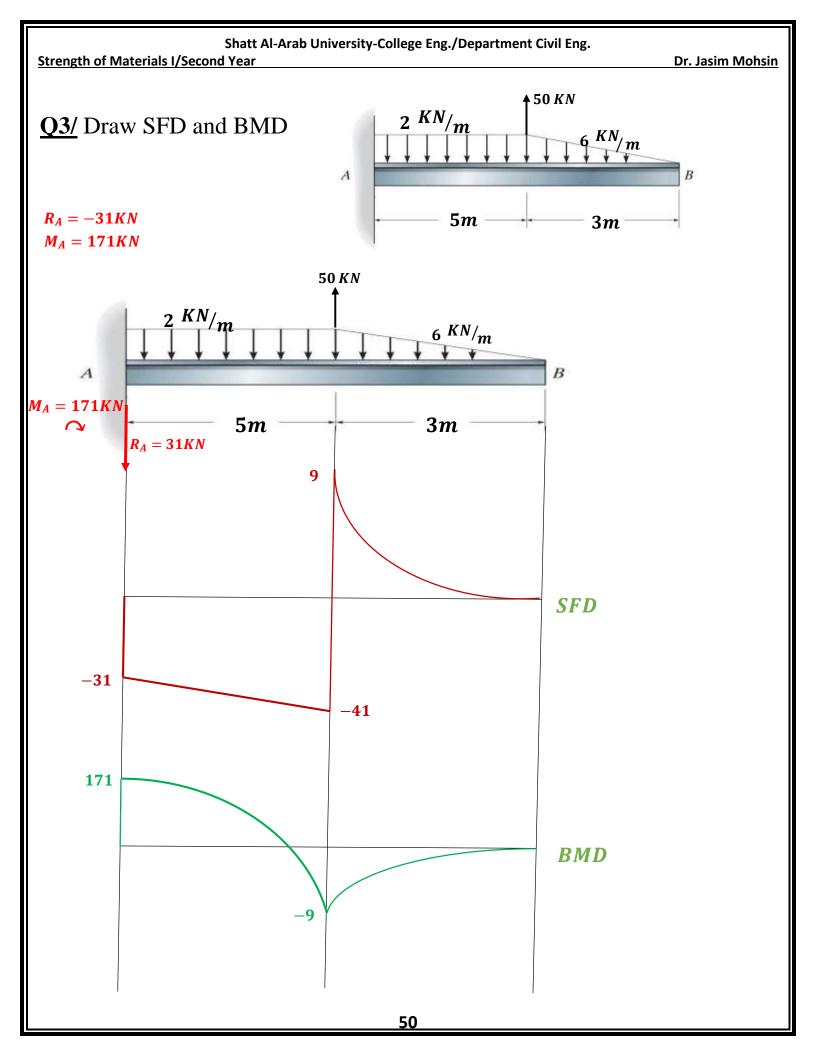


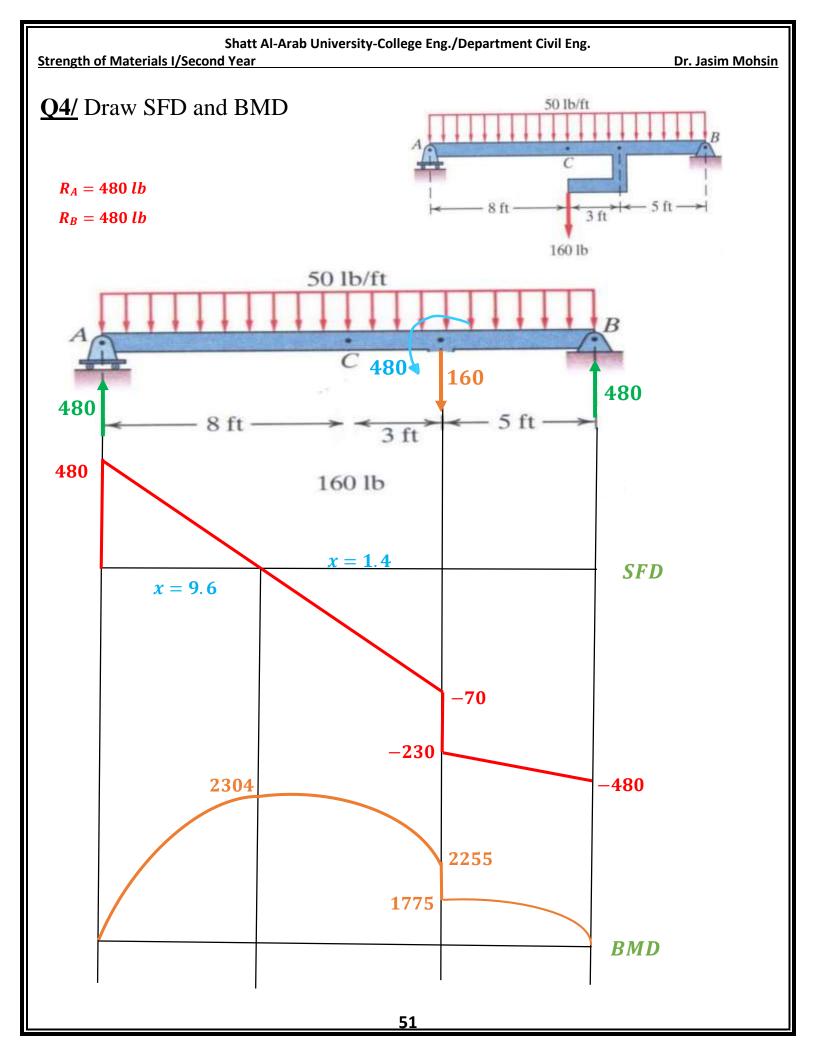


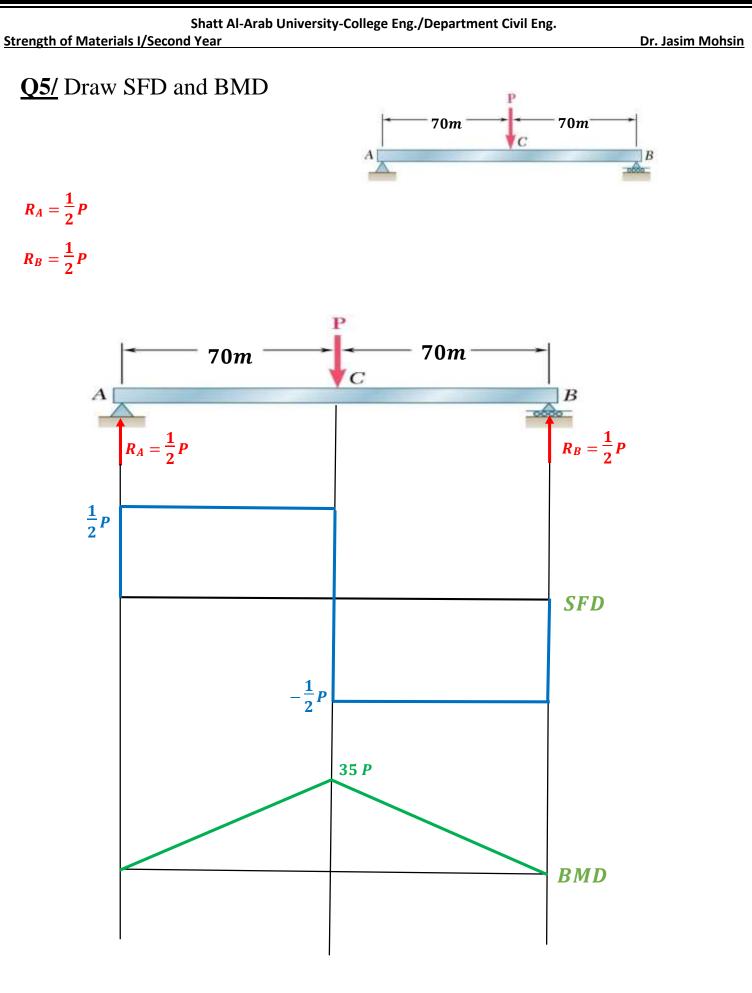


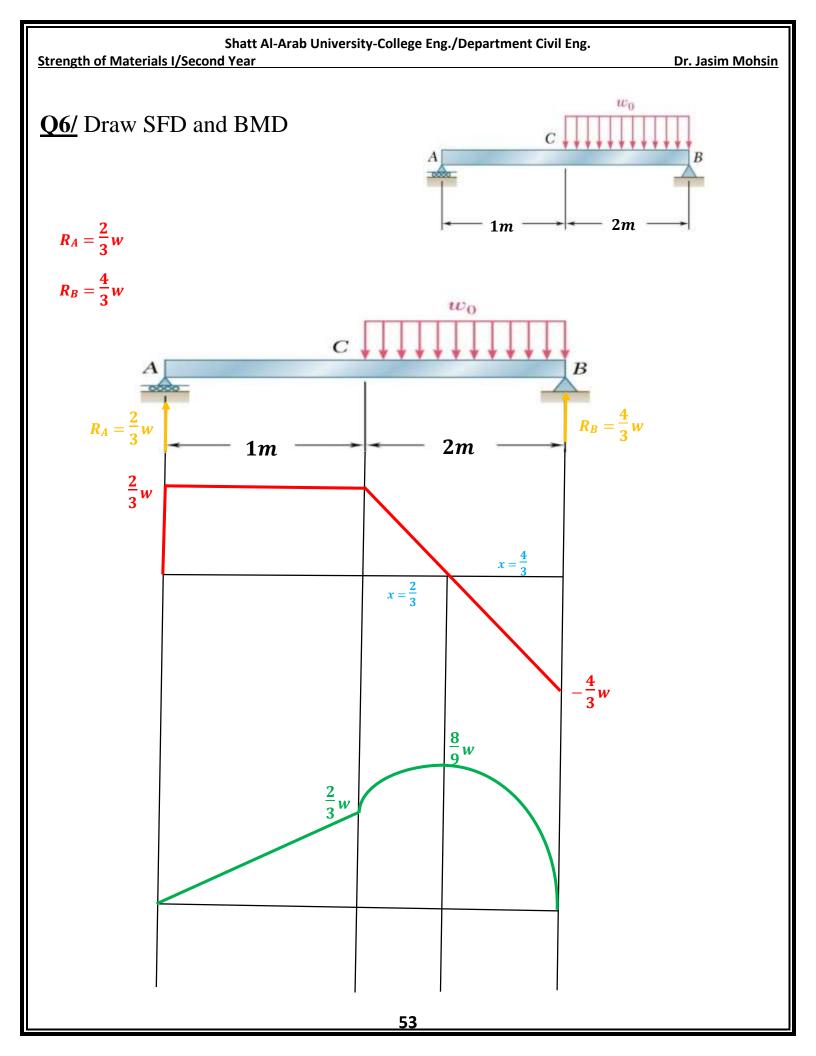












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<u>7. Chapter Seven: Thin Walled Cylinder</u> <u>Thin – walled Cylinder</u>



1- Hoop stress (σ_h)



Circumferential

2– longitud stress (σ_l)

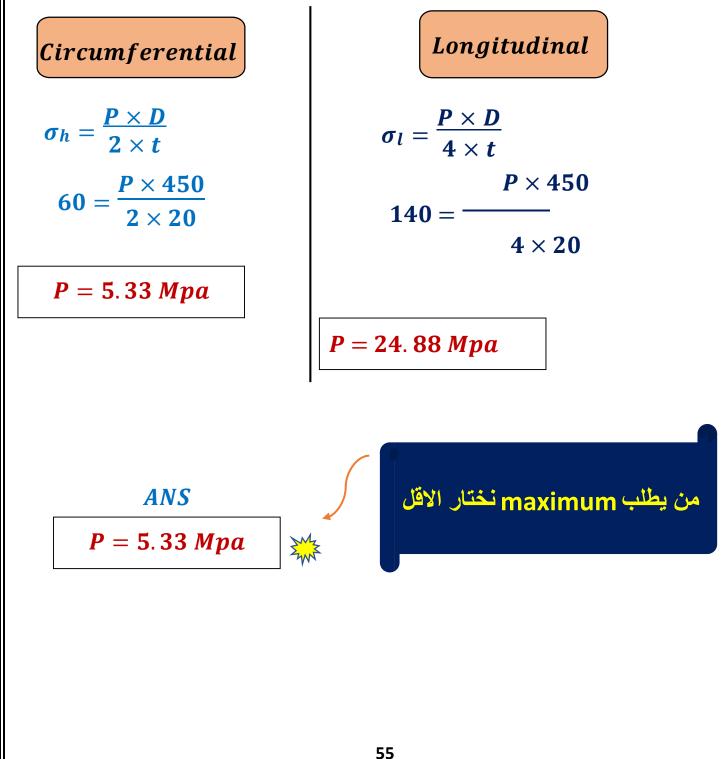
$$\sigma_{I} = \frac{P \times D}{4 \times t}$$

$$\sigma_{h} = \frac{P \times D}{2 \times t}$$

$$\sigma_{h} = \frac{P \times D}{2 \times t}$$



Q1/ A cylindrical pressure vessel is fabricated from steel plating that has a thickness of 20 mm. The diameter of the pressure vessel is 450 mm and its length is 2.0 m. Determine the maximum internal pressure that can be applied if the longitudinal stress is limited to140 MPa, and the circumferential stress is limited to 60 MPa



Strength of Materials I/Second Year

Dr. Jasim Mohsin

<u>**O2**</u>/A water tank is 8m in diameter and 12m high . If the tank is to be completely filled , determine the minimum thickness of the tank plating if the stress is limited to 40Mpa ? The density of water is 1000Kg / m³

$$P = \rho \times g \times h$$

$$P = 1000 \times 10 \times 12$$

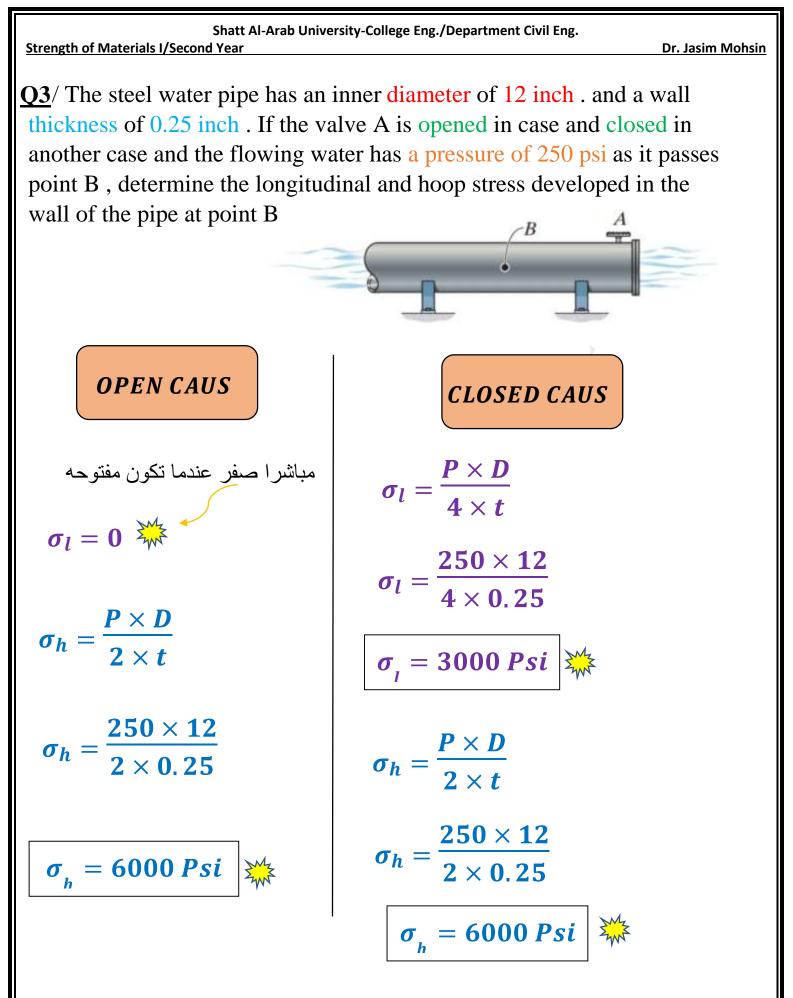
$$P = 120000 Pa$$

$$\sigma_{h} = \frac{P \times D}{2 \times t}$$

$$t = 12 mm$$

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$$t = 12 mm$$



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Q4 / A cylindrical pressure vessel is fabricated from steel plates which have a thickness of 20mm. The diameter of the pressure vessel is 500mm and it's length is 3m. Determine the maximum internal pressure Which can be applied if the stress in the steel is limited to 140 Mpa. If the internal pressure were increased until The vessel burst, sketch the type of fracture which would occur



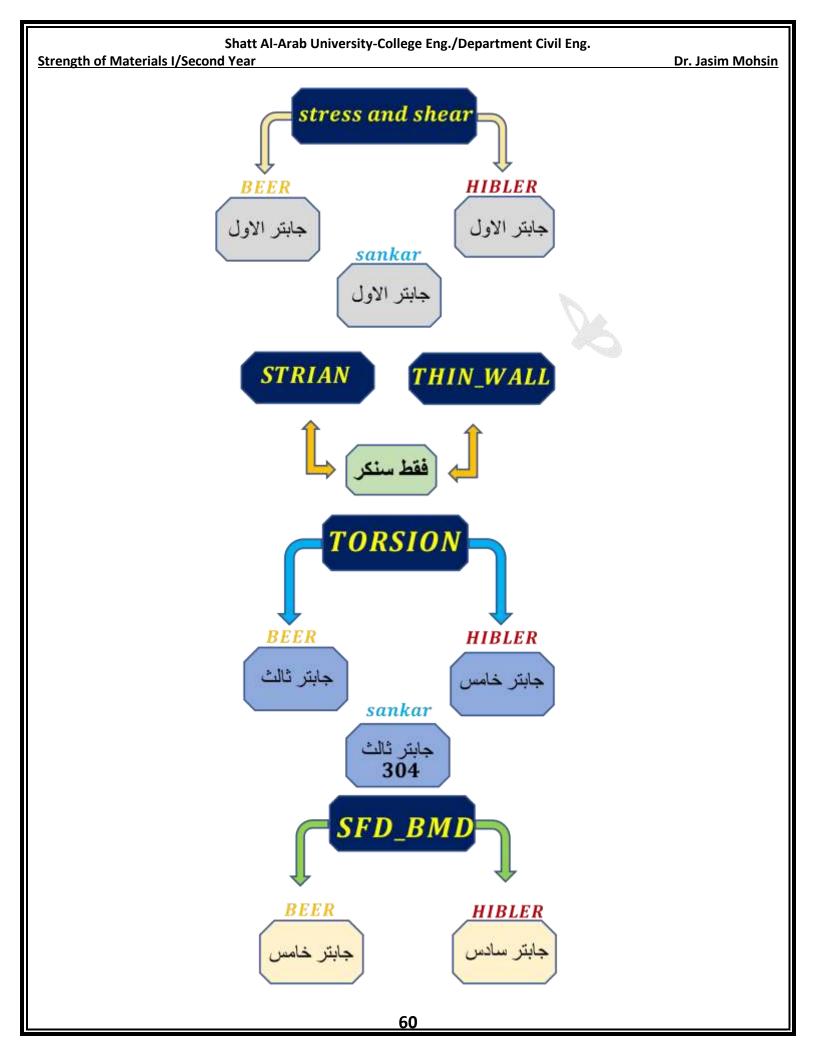


Strength of Materials I/Second Year Dr. Jasim Mohsin **Q5**/ The tank shown in figure, is fabricated from 10mm steel plate. Determine the maximum longitudinal and circumferential (hoop) stresses caused by an internal pressure of 1.2Mpa 600mi $\sigma_h = \frac{P}{A} = \frac{PA}{2tl} = \frac{PDl}{2tl}$ $F = \mathbf{1} \cdot \mathbf{2} \times (600 \times 400 + \frac{\pi}{4} \times 400^2)$ $\sigma_h = \frac{PD}{2t}$ F = 438796N $\sigma_l = \frac{P}{A} = \frac{P}{2wt + \pi dt}$ $\sigma_h = \frac{1.2 \times (600 + 400)}{2 \times 10}$ $\sigma_l = \frac{438796}{2 \times 600 \times 10 + \pi \times 400 \times 10}$ $\sigma_h = 60Mpa$

F

 $\sigma_{l} = 17.86 Mpa$

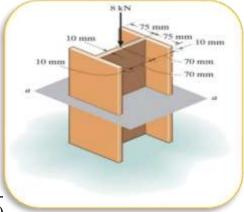
F = PA



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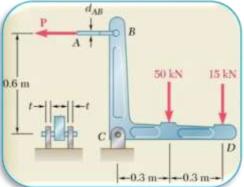
1)The column is subjected to an axial force of 8 kN, which is applied through the centroid of the cross- sectional area. Determine the average normal stress acting at section a - a . Show this distribution of stress acting over the area's cross section



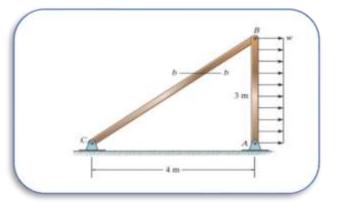
2)Two forces are applied to the bracket BCD as shown. (a) Knowing that the control rod AB is to be made of a steel having an ultimate normal stress of 600 MPa, determine the diameter of the rod for which the factor of safety with respect to failure will be 3.3. (b) The pin at C is to be made of a steel having an ultimate shearing stress of 350 MPa. Determine the diameter of the pin C for which the factor of safety with respect to shear will also be 3.3. (C) Determine the required thickness of the bracket supports at C knowing that the allowable bearing stress of the steel used is 300 MPa

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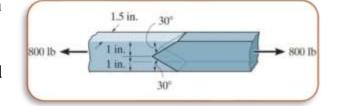
3)Determine the largest intensity w of the uniform loading that can be applied to the frame without causing either the average normal stress or the average shear stress at section b- b to $\sigma = 15$ MPa and $\tau = 16$ MPa, respectively. Member CB has a square cross section of 30 mm on each side



4)The two members used in the construction of an aircraft fuselage are joined together using a 30 ° fish - mouth weld . Determine the average normal and average shear stress on the plane of each weld . Assume each inclined . plane supports a

horizontal force of 400 lb

Ans w = 20000N





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5)A 200- mm - diameter pulley is prevented from rotating relative to 60 - mm - diameter shaft by a 70 - mm - long key, as shown If a torque T = 2.2 kN - m is applied to the shaft, determine the width b if the allowable shearing stress in the key is 60 MPa

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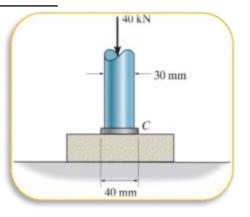
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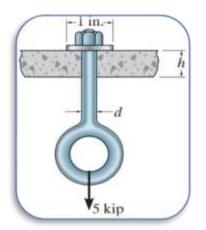
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Shaft 50 mm-d

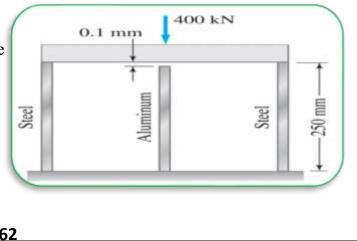
6)The shaft is subjected to the axial force of 40 kN . Determine the average bearing stress acting on the collar C and the normal stress in the shaft

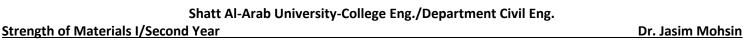


7) The eye bolt is used to support the load of 5 kip. Determine its diameter d to the nearest $\frac{1}{8}in$. and the required thickness h to the nearest $\frac{1}{8}in$. of the support so that the washer will not penetrate or shear through it. The allowable normal stress for the bolt is $\sigma_{allow} = 21$ ksi and the allowable shear stress for the supporting material is $\sigma_{allow} = 5$ ksi



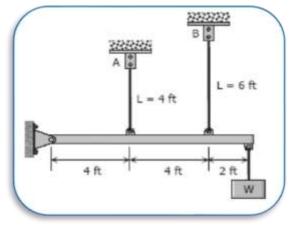
8)Before the P= 400 kN load is applied, the rigid platform rests on two steel bars, each of cross sectional area 1200 mm², as shown in the figure . The cross - sectional area of the aluminum bar is 2400 mm². Compute the stress in the aluminum bar after the 400 kN load is applied. Use E = 200 GPa for steel and E = 70 GPa for aluminum. Neglect the weight of the platform





9) The two vertical rods attached to the light rigid bar are identical except for length. Before the load W was attached, the bar was horizontal and the rods were stress - free . Determine the load in each rod if W = 6600 lb





10) The light rigid bar ABCD shown is pinned at B and connected to two vertical rods . Assuming that the bar was initially horizontal and the rods stress - free, determine the stress in each rod after the load after the load P = 20 kips is applied

11)Two cylindrical rods one of steel and the other of brass, are joined at C and restrained by rigid supports at A and E. For the loading shown and knowing that E, 200 GPa and E = 105 GPa, deter mine (a) the reactions at A and E, (b) the deflection of point C

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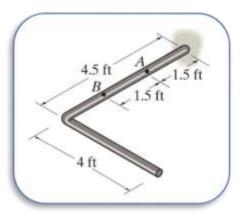
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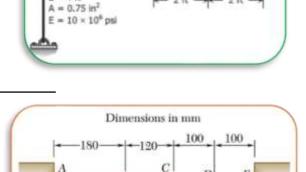
Steel B . 60 kN 40-mm diam.

= 4 ft

12) The rod has a diameter of 1 in . and a weight of 15 lb / ft. Determine the maximum torsional stress in the rod at a section located at B due to the rod's weight

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Brass .

30-mm diam.

40 kN

0.5 in² 29 × 10⁴ pp

